

**The Preignition and Autoignition Oxidation of Alternatives to Petroleum
Derived JP-8 and their Surrogate Components in a Pressurized Flow Reactor
and Single Cylinder Research Engine**

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Dedication

This work is dedicated to my family, friends, and Jolly.

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Table of Contents

LIST OF TABLES	viii
LIST OF FIGURES	ix
ABSTRACT	xi
CHAPTER 1: INTRODUCTION	1
1.1 Motivation	1
1.2 Research Objective	3
1.3 Research Program	3
CHAPTER 2: BACKGROUND AND LITERATURE REVIEW	6
2.1 Overview	6
2.2 Hydrocarbon Oxidation	6
2.3 Chemistry of Small Alkanes	9
2.4 Chemistry of n-Decane	11
2.5 Chemistry of Cycloalkanes	13
2.6 Chemistry of Petroleum and Alternative Jet Fuels	15
2.6.1 Petroleum JP-8	16
2.6.2 Fischer-Tropsch Jet Fuel	17
2.6.3 Coal Derived Jet Fuel	19
2.6.4 Blends of Alternative Jet Fuels	20
2.7 Closure	21
CHAPTER 3: EXPERIMENTAL FACILITY	23
3.1 Introduction	23
3.2 Pressurized Flow Reactor Facility	24
3.3 Single Cylinder Research Engine Facility	25

3.4 Controlled Cool Down	28
3.5 Siemens Ultramat 22P CO and CO ₂ Detector.....	28
3.6 Sample Storage System.....	29
3.7 Trace GC Gas Chromatograph and Trace DSQ Mass Spectrometer ..	32
3.8 Pressurized Flow Reactor Facility Upgrades and Maintenance	38
3.8.1 Sample Probe Construction and Laser Welding	38
3.8.2 Replacement of the Inlet Bead Heater Controller	40
3.8.3 Replacement of the Chromalox 10 kW Air Circulation Heater and Controllers	40
3.8.4 Cleaning the 3 kW Heater Electrical Connectors	42
3.9 Closure	44
CHAPTER 4: FISCHER-TROPSCH JET FUEL, PETROLEUM JP-8, AND N- DECANE/ISO-OCTANE OXIDATION	45
4.1 Introduction.....	45
4.2 Results and Discussion	47
4.2.1 CO and CO ₂ Comparison of Petroleum JP-8 with Fischer-Tropsch Jet Fuel	47
4.2.2 Oxidation of the n-Decane/iso-Octane Mixture	49
4.2.3 Oxidation of the Tuned n-Decane/iso-Octane Mixture.....	50
4.3 Closure	51
CHAPTER 5: THE OXIDATION OF ALTERNATIVE JET FUELS AND A FISCHER-TROPSCH JET FUEL SURROGATE IN COMPLEMENTARY COMBUSTION FACILITIES.....	53
5.1 Introduction.....	53
5.2 Experimental Conditions	54
5.3 Results and Discussion	55
5.3.1 Jet Fuels in the PFR.....	55

5.3.2 Petroleum JP-8, Coal Derived Jet Fuel, and Fischer-Tropsch Jet Fuel Surrogate in the Single Cylinder Research Engine	58
5.4 Closure	61
CHAPTER 6: INTERMEDIATE SPECIES ANALYSIS OF N-DECANE AND DECALIN	63
6.1 Introduction.....	63
6.2 Experimental Conditions	64
6.3 Results and Discussion	64
6.3.1 Detailed Speciation of Intermediates Produced during n-Decane Oxidation in the PFR.....	65
6.3.2 Detailed Speciation of Intermediates Produced during Decalin Oxidation in the PFR.....	69
6.4 Closure	75
CHAPTER 7: CONCLUSIONS AND RECOMMENDATIONS FOR FUTURE WORK	77
7.1 Introduction.....	77
7.2 Conclusions.....	78
7.3 Recommendations for Future Work.....	80
7.4 Closure	82
LIST OF REFERENCES	83
APPENDIX A: GENERAL OPERATING PROCEDURE FOR THE PRESSURIZED FLOW REACTOR DURING A CONTROLLED COOL DOWN EXPERIMENT...	89
A.1 Typical PFR Heater Set-Point Schedule	98
A.2 Pictures of Selected PFR Facility Components	100

LIST OF TABLES

Table 1.1: JP-8 specifications defined by MIL-DTL-83133E.	3
Table 2.1: Comparison of selected properties of the jet fuel samples.	15
Table 2.2: Composition of petroleum JP-8, POSF 3773.	17
Table 2.3: Composition of Fischer-Tropsch Jet Fuel, POSF 4734.	18
Table 2.4: Components in Fischer-Tropsch Jet Fuel, POSF 4734.	18
Table 2.5: Composition of coal derived jet fuel, POSF 4765.	20
Table 3.1: Selected research engine specifications.	27
Table 3.2: Typical experimental operating conditions of the PFR.	28
Table 3.3: MS operating parameters.	33
Table 3.4: Resistance of each heater element.	43
Table 4.1: Properties of petroleum and Fischer-Tropsch samples.	46
Table 4.2: Properties of the mixture and components.	46
Table 5.1: Properties of coal derived and Fischer-Tropsch jet fuel samples.	54
Table 6.1: Species identified by GC/MS/FID in n-decane oxidation samples.	68
Table 6.2: Species identified by GC/MS/FID in decalin oxidation samples.	71
Table A.1: Typical set-points for PFR heaters during warm-up phase.	99

LIST OF FIGURES

Figure 2.1: Branching pathways of hydrocarbon oxidation at low and intermediate temperatures. Based on extensive experimental and modeling research of linear and lightly branched paraffins of C ₃ – C ₈ by many groups.....	8
Figure 3.1: Schematic of the PFR identifying key components.	25
Figure 3.2: Schematic of the single cylinder research engine facility with key components identified.	27
Figure 3.3: Schematic of the heated sample storage system cart in sample collection mode.	31
Figure 3.4: Schematic of the heated sample storage system cart in sample analysis mode.	32
Figure 3.5: Programmed GC temperature profile.	33
Figure 3.6: 4-port and 6-port valves in the off position, prep run.	35
Figure 3.7: GC valves in the sample injection position with the 6-port valve on and the 4-port valve off.	36
Figure 3.8: 4-port and 6-port valves off in the sample injection position.....	36
Figure 3.9: Position of GC valves at 4.00 minutes with the 6-port valve off and the 4-port valve on.	37
Figure 3.10: Off position of the 4-port and 6-port valves at end of program.	37
Figure 3.11: Schematic of the sample probe tip showing laser welding locations and dimensions.....	39
Figure 3.12: Wiring diagram of 3 kW heater elements. Connection e and f are on the backside of the heater.....	43
Figure 4.1: Chemical structure of each species in the binary mixture.....	47
Figure 4.2: CO production of Fischer-Tropsch jet fuel and petroleum JP-8.	48
Figure 4.3: CO ₂ production of Fischer-Tropsch jet fuel and petroleum JP-8.....	49
Figure 4.4: CO and CO ₂ production of 59.4% n-decane/40.6% iso-octane mixture..	50
Figure 4.5: Reactivity map of Fischer-Tropsch jet fuel and 53.1% n-decane / 46.9% iso-octane mixture.	51

Figure 5.1: Reactivity map of natural gas, petroleum, and coal derived jet fuel in the pressurized flow reactor.	57
Figure 5.2: Autoignition of natural gas, petroleum, and coal derived jet fuel in the single cylinder research engine.	59
Figure 5.3: Autoignition of surrogate for Fischer-Tropsch jet fuel.	61
Figure 6.1: Reactivity of n-decane, Fischer-Tropsch jet fuel, and n-decane/iso-octane blend based on CO production.	66
Figure 6.2: CO and CO ₂ production during n-decane oxidation.	67
Figure 6.3: Chromatogram of the 700 K n-decane oxidation sample.	68
Figure 6.4: Chemical structure of the identified species during n-decane oxidation..	69
Figure 6.5: Reactivity map of decalin oxidation in the PFR.....	70
Figure 6.6: Chemical structure of the identified species during decalin oxidation. ...	72
Figure 6.7: MS (top) and FID (bottom) chromatograms produced from decalin oxidation at 690 K.	74
Figure A.1: Picture of PFR controllers with arrows identifying main power supply control switches.....	100
Figure A.2: Picture of PFR computer with the arrow identifying the PFR mass flow controller.	101
Figure A.3: Picture of sample connections to the CO and CO ₂ detector, and calibration controls.....	102
Figure A.4: Picture of air operated sample valve identified with arrow.....	103
Figure A.5: Picture of the ISCO Model 500D fuel syringe pump.	103

ABSTRACT

The Preignition and Autoignition Oxidation of Alternatives to Petroleum Derived JP-8 and their Surrogate Components in a Pressurized Flow Reactor and Single Cylinder Research Engine

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The U.S. Department of Defense (DoD) Directive 4140.13 has required the use of JP-8 specification fuels whenever possible. The specifications for JP-8 allow a broad range of thermophysical characteristics, such as chemical composition, distillation characteristics, and heat of combustion, which can be produced from multiple sources and processes. For applications in military power systems operating with compression ignition (CI) engines and some gas turbines, it is necessary to study the effects of fuel source on the preignition and autoignition reactivity that occurs in the 600-1000 K temperature range of the resulting jet fuels and to develop reactivity surrogates for their behavior. The use of alternative sources for JP-8, rather than petroleum sources, has these advantages: improved U.S. energy security, a wider range of feedstocks, and decreased emissions of SO_x and particulate matter. To study the effect of fuel source, the oxidation of three jet fuels from different sources (petroleum, natural gas, and coal) was examined. Preignition experiments were conducted in a pressurized flow reactor (PFR) under lean, dilute conditions at temperatures of 600-800 K and 8 atm pressure. Autoignition experiments were conducted in a single cylinder research engine. Results were compared to a typical sample of petroleum-derived jet fuel.

Petroleum, natural gas, and coal derived fuels contain hundreds of components, many with unknown reactivity, and even if known, proper kinetic

simulations require computational resources beyond current abilities. To overcome these limitations, the use of surrogates, mixtures of approximately 1-10 components that mimic the properties and behaviors of the real fuel, has been recognized as a feasible approach to chemical kinetic model building. Since the chemical composition varies widely between petroleum and alternative jet fuels, new surrogates need to be developed for alternative fuels as well.

Samples of the jet fuels derived from coal, natural gas, and petroleum were oxidized in two complementary experimental facilities to explore their preignition and autoignition behavior. In both facilities, the order of reactivity, based on carbon monoxide production, in descending order was Fischer-Tropsch, petroleum, and coal. The reactivity differences are attributed to composition differences. Results showed that all the fuels exhibit negative temperature coefficient behavior as expected, but Fischer-Tropsch jet fuel produces significantly more carbon monoxide than petroleum and coal derived jet fuel before entering the negative temperature coefficient region.

Possible surrogates and their components were also tested in the facilities to elucidate how compositional differences affect preignition and autoignition chemistry. A mixture of n-decane/iso-octane was studied as a surrogate fuel for Fischer-Tropsch jet fuel. The mixture was then tuned to approximate the low and intermediate temperature reactivity of Fischer-Tropsch jet fuel. Fischer-Tropsch jet fuel and the mixture showed similar reactivity based on measurements of carbon monoxide.

To further explore the chemistry and impact of the n-decane component of the surrogate for Fischer-Tropsch jet fuel and the decalin component of coal derived jet fuel, PFR experiments were conducted to identify intermediate species profiles using gas chromatography with flame ionization detection and gas chromatography coupled to a mass spectrometer. The major intermediate species that were identified with n-decane oxidation included components of the functional groups aldehydes and alkenes. Cyclic species were the major species observed with decalin oxidation. The research presented will aid in the overall development and use of chemical kinetic models that will be employed for simulating combustion characteristics of gas turbines and CI engines while improving such traits as fuel efficiency, emissions, and power output.

CHAPTER 1: INTRODUCTION

1.1 Motivation

While there are replacements for hydrocarbon fuels in many energy sectors, there are no candidates to replace the energy density of liquid fuels for air transportation and long haul freight movers. As the world approaches the production maximum from petroleum reserves, there exists a need to produce alternatively sourced liquid fuels, defined as fuels from non-petroleum sources. If the raw material for such fuels is discovered in the U.S., their use would increase U.S. energy security. Many such replacements exist: for example, a jet fuel using Fischer-Tropsch processing of syngas (CO and H_2) generated from natural gas (Muzzell et al., 2006) or a Pennsylvania State University, University of Dayton, PARC Technical Services, and Wright-Patterson Air Force Base coal-derived jet fuel manufactured in a hydrotreating process (Balster et al., 2008), to name two. Both fuel samples meet most U.S. military jet fuel JP-8 specifications and could probably be used as the international commercial jet fuel, Jet A-1, but with the addition of a fuel system icing inhibitor, a corrosion inhibitor, and a static dissipater for JP-8 applications.

Though envisioned as a jet fuel, the DoD has designated JP-8 as the single fuel forward for all combustion engines wherever applicable (TARDEC, 2001). This includes running JP-8 in compression ignition (CI) engines, where the preignition and autoignition chemistry at temperatures between 600 and 1000 K can play a crucial role. Combustion chemistry at these temperatures is a complex process involving competing reaction paths and noted by phenomena such as cool flames and Negative Temperature Coefficient (NTC) behavior.

As simulations have become more integral to the design process, chemical kinetic models for JP-8 are needed for the design of both next generation jet and CI engines. Because of its broad composition, direct development of chemical kinetic models for JP-8 is not possible. Instead, utilization of appropriate surrogates, mixtures of approximately 1-10 components that mimic the properties and behavior of the real fuel, have been suggested as a feasible approach to model building (Colket et al., 2007).

JP-8 composition specifications are based on general physical properties rather than specific compositions, as shown in Table 1.1, to allow its production from alternative feedstocks and with a variety of processing methods. This practice leads to wide variations in JP-8 composition. JP-8 derived from petroleum contains significant amounts of n- and iso-paraffins, mono- and dicycloparaffins, and alkylated aromatics. Jet fuel produced through a Fischer-Tropsch process is composed entirely of n- and iso-paraffins. Coal-derived jet fuel produced through hydrotreating produces a fuel of 97% mono-, di-, and tricycloparaffins (Balster et al., 2008).

The use of alternative sources for JP-8 rather than petroleum derived fuels has these advantages: improved U.S. energy security, a wider range of feedstocks, and decreased emissions of SO_x and particulate matter. U.S. energy security can be improved with the aid use of resources found within the U.S. If resources such as coal and natural gas can be used to produce acceptable liquid fuels, it reduces the need to rely on foreign nations for petroleum. Harmful emissions, such as SO_x and particulate matter, are reduced from the Fischer-Tropsch and coal-derived jet fuel as they contain no sulfur or aromatic components.

Table 1.1: JP-8 specifications defined by MIL-DTL-83133E.

Property	Minimum	Maximum
Aromatics	-	25.0% vol
Alkenes	-	5.0% vol
Naphthalenes	-	3.0% vol
Total Sulfur	-	0.30% mass
Distillation – 10% Recovered	-	205 °C
Evaporation Point	-	300 °C
Flash Point	38 °C	-
API Gravity	37.0	51.0
Freezing Point	-	-47 °C
Heat of Combustion	42.8 MJ/kg	-
Hydrogen Content	13.4% mass	-
Fuel System Icing Inhibitor	0.10% vol	0.15% vol

1.2 Research Objective

The objective of this study was to compare the low and intermediate temperature oxidation characteristics of petroleum derived jet fuel and jet fuel produced from alternative sources and processes. The experimental data will aid in the development and validation of chemical kinetic mechanisms which will be useful in designing new and improved engines.

1.3 Research Program

The experimental approach of this study was to oxidize a selected jet fuel, under lean, dilute conditions at temperatures of 600-800 K (low and intermediate temperature regime) and 8 atm pressure in a pressurized flow reactor and single cylinder research engine. There was also an attempt to develop a surrogate fuel for low and intermediate temperature reactivity. Chapters in this thesis are based on presented papers and papers in preparation for journal submission. The specific experimental stages are listed below:

(1) Examine the reactivity of a natural gas based Fischer-Tropsch jet fuel manufactured by Syntroleum and designated as S-8. Compare reactivity to that of petroleum-derived JP-8 in the pressurized flow reactor, and propose surrogate components for Fischer-Tropsch jet fuel. The program included: (a) determining the reactivity of the fuels by measuring the CO produced over the low and intermediate temperature regime and examining the similarities and differences of the alternative and petroleum fuels with respect to CO production; (b) choosing a mixture of hydrocarbon components that include linear and branched alkanes to represent the average molecular formula of Fischer-Tropsch jet fuel and to approximate a cetane number slightly higher than that of jet fuel, in anticipation of the higher reactivity of Fischer-Tropsch jet fuel; and (c) after choosing a mixture of n-decane and iso-octane as surrogate components, performing experiments with the pressurized flow reactor while varying the cetane number of the surrogate mixture to match reactivity of a sample of Fischer-Tropsch jet fuel. This particular phase of the study was the first investigation of an alternative jet fuel in the pressurized flow reactor at Drexel University.

(2) Further explore the n-decane/iso-octane mixture as a surrogate for Fischer-Tropsch jet fuel. The experimental method consisted of: (a) tuning the binary mixture composition to approximate the reactivity of Fischer-Tropsch jet fuel throughout the low and intermediate temperature regime; and (b) measuring the intermediate species produced during the low and intermediate temperature oxidation of neat n-decane with the aid of gas chromatography and mass spectrometry, in order to identify the branching pathways controlling auto-ignition. n-Decane was studied

neat, rather than in a n-decane/iso-octane mixture, in order to simplify the investigation of the branching pathways produce from n-alkane oxidation.

(3) Investigate and compare the reactivity of petroleum derived, natural gas derived Fischer-Tropsch, and coal-derived jet fuel in both the pressurized flow reactor and a single cylinder research engine. Decalin is the most prevalent component of coal-derived jet fuel. In order to determine important pathways for the oxidation of decalin, decalin was oxidized in the flow reactor and samples were extracted for analysis. Ultimately, the pathways will aid in the improvement and advancement of the chemical mechanism for decalin in the low and intermediate temperature regime at elevated pressure. The overall experimental method consisted of: (a) exploring and comparing the reactivity of each of the jet fuels in both facilities; (b) examining the n-decane/iso-octane mixture in the single cylinder research engine to test the oxidation behavior and to compare the reactivity to that of the Fischer-Tropsch jet fuel; (c) collecting and performing a detailed chemical analysis of decalin PFR samples with the aid of gas chromatography and mass spectrometry. This particular phase of the study was the first time that alternative fuels, natural gas and coal derived jet fuel, were examined in the single cylinder research engine at Drexel University.

CHAPTER 2: BACKGROUND AND LITERATURE REVIEW

2.1 Overview

This chapter aims to provide an overview of the present state of our understanding of hydrocarbon combustion in the low and intermediate temperature regimes. Additionally, this chapter describes past and ongoing combustion research related to n-alkanes, cycloalkanes, and alternative jet fuels. Section 2.2 introduces details pertaining to the chemistry of hydrocarbon oxidation and the theory of chemical kinetics. Section 2.3 discusses kinetics of small alkanes. Section 2.4 discusses details of previous n-decane work. Previous work concerning cycloalkanes is described in Section 2.5. Section 2.6 discusses petroleum and alternative jet fuels. Section 2.7 presents the closure to Chapter 2.

2.2 Hydrocarbon Oxidation

The oxidation of saturated, straight chain hydrocarbons is characteristically divided into three temperature regimes:

- 1.) Low temperature (< 650 K),
- 2.) Intermediate temperature (650-1000 K), and
- 3.) High temperature (> 1000 K).

The temperature boundaries correspond to atmospheric pressure, and shift to higher temperatures with increasing pressure. Different reaction pathways are dominant at different temperatures, and the pathways are controlled by the concentration of radical species. At low temperatures, alkylperoxy radicals ($\text{RO}_2\bullet$) are dominant. Hydroperoxy ($\text{HO}_2\bullet$) and hydroxyl ($\text{OH}\bullet$) radicals control the intermediate

temperature regime, and $\text{OH}\bullet$, oxygen ($\text{O}\bullet$), and hydrogen ($\text{H}\bullet$) radicals dominate the high temperature regime. Hydrocarbons with other structures may or may not behave accordingly to this hierarchy. Therefore, chemical composition affects reactivity behavior of hydrocarbons. Each of the temperature regimes come into play in CI engines.

Modern explanations of hydrocarbon oxidation are based on the mechanism of free radicals proposed by Semenov (1935). The general theory of autoignition for smaller n-paraffins is well-accepted based on extensive research and modeling (e.g., Miller et al., 2005; Law, 2006). Figure 2.1 shows the general scheme of linear and lightly branched paraffins with carbon number 3 and higher. Initially, hydrogen is abstracted from the parent fuel molecule RH to produce the $\text{R}\bullet$ alkyl radical. The formation of the initial radical from the parent fuel molecule is known as primary initiation. Secondary initiation occurs when a radical is formed from a stable species other than the parent fuel molecule. Oxygen addition to the alkyl radical will produce the alkylperoxy radical $\text{RO}_2\bullet$. The alkylperoxy radical has a number of different possible pathways (e.g., decomposition to produce alkoxy radicals, $\text{RO}\bullet$, and aldehydes RCHO , reversible isomerization to produce an alkylhydroperoxy radical, $\text{Q}\bullet\text{OOH}$, and reaction with a fuel molecule to produce an alkyl radical, $\text{R}\bullet$). The reaction of $\text{RO}_2\bullet$ with a fuel molecule yields one alkyl radical, $\text{R}\bullet$, and alkylhydroperoxide, ROOH . Reactions that convert one radical into another radical are known as chain propagation reactions. If the temperature is high enough, the alkylhydroperoxy may decompose by beta scission to produce an alkene, a carbonyl, and a hydroxyl, or react with O_2 to produce a dihydroperoxide. This may decompose

to produce a ketohydroperoxide and a hydroxyl. Further decomposition will produce another hydroxyl, a carbonyl, and an aldehyde. If the temperature increases to approximately 850 K, the intermediate temperature region reactions will become important, and the alkyl radical can react with O_2 to produce a hydroperoxy radical, $HO_2\bullet$, and an alkene. The $HO_2\bullet$ may abstract an H-atom from the parent fuel molecule producing hydrogen peroxide. Decomposition of hydrogen peroxide, when the temperature is high enough, produces 2 hydroxyls. The conversion of 1 radical into 2 radicals is called a chain branching pathway. Hydroxyls react with parent fuel molecules to produce alkyl radicals and H_2O . Reactions which reduce the number of radicals in the system are called termination reactions.

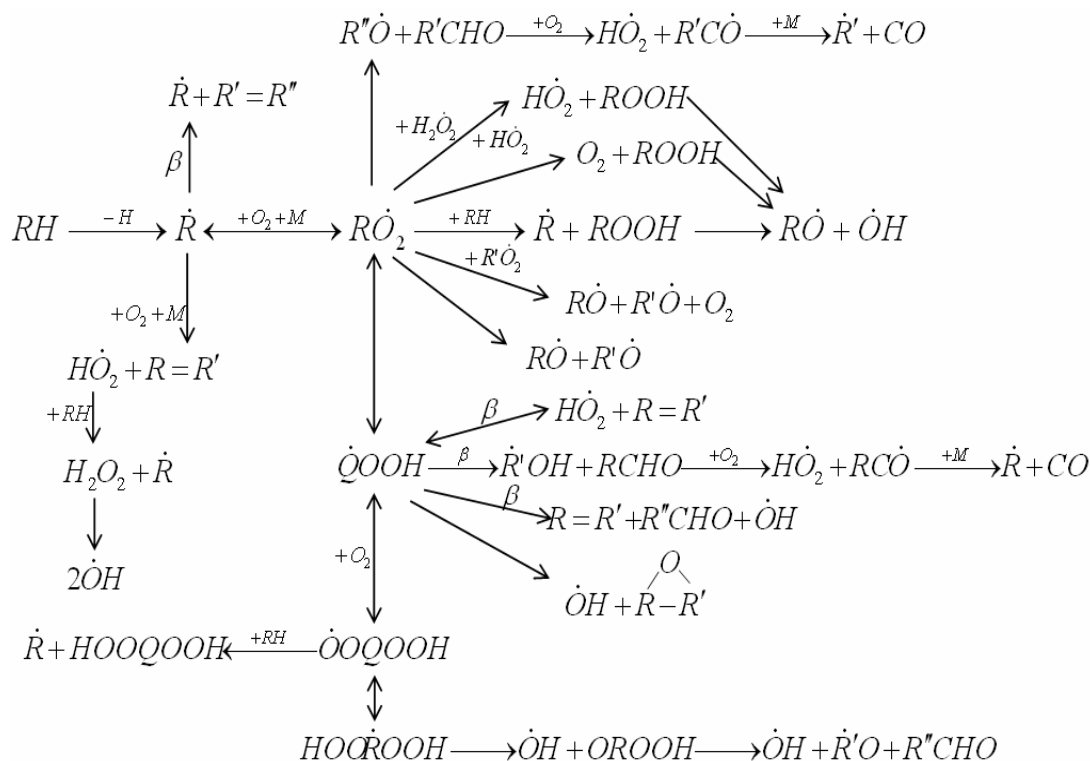


Figure 2.1: Branching pathways of hydrocarbon oxidation at low and intermediate temperatures. Based on extensive experimental and modeling research of linear and lightly branched paraffins of $C_3 - C_8$ by many groups.

Clearly, hydrocarbon oxidation chemistry that occurs in the low and intermediate temperature regime is a complex process that contains numerous competing reactions. To add to the complexity, as the molecular weight of the hydrocarbon increases, the number of possible intermediate species produced from the different reaction pathways at these temperatures also increases. To further improve and expand chemical kinetic models, gas chromatography and mass spectroscopy can be utilized to identify and quantify intermediate species produced and to classify which reaction pathways are dominant.

2.3 Chemistry of Small Alkanes

Alkanes represent the most abundant chemical class found in JP-8 jet fuel (Edwards et al., 2001). This section is a brief report of selected work dealing with alkanes lighter than n-decane. Propane has been previously studied with the pressurized flow reactor (Koert et al., 1994). Propane was oxidized between 600-900 K, with lean equivalence ratio of 0.4, and elevated pressure of 10 and 15 atm. Low temperature reactivity, as a measure of CO production, occurred during 680-770 K. Maximum production of several species produced during the oxidation occurred at approximately 720 K. CO was the major species that was produced with a peak production at 780 K. Propane was also studied in a jet stirred flow reactor with temperatures of 900-1200 K, pressures ranging from 1-10 atm, and a wide range of equivalence ratios (Dagaut et al., 1987). There has been rapid compression machine work with n-butane between 700-900 K, a wide range of equivalence ratios from 0.8-1.2, and elevated pressures from 9-11 bar (Minetti et al., 1994). Products of combustion were also measure with the use of a gas chromatograph coupled to a mass

spectrometer. Results indicated that 3 different distinct temperature zones can be identified. The temperature zone between 700-760 K had two-stage ignition characteristics such as pressure buildup and light emission from a cool flame. The 760-850 K temperature zone had a increase in total delay time with increasing temperature to signify the negative temperature coefficient regime. From 850 K and higher, ignition occurs in one stage and the delay time decreases. Major carbon containing species that were identified and quantified throughout zones included 1-butene, methanol, carbon monoxide, 2-butene, and ethene. A noticeable similarity between each of the products with respect to two-stage ignition was observed (Minetti et al., 1994).

Since it is a primary reference fuel for gasoline, iso-octane (2,2,4-trimethylpentane) has received considerable attention. A modeling study by Curran et al. (2002) reviewed experimental data over the temperatures 550-1700 K and pressures 1-45 atm and developed an iso-octane model. The kinetic model for iso-octane has been modified and updated (Tanaka et al., 2003; He et al., 2005; Jia et al., 2006; Chaos et al., 2007) as new data have become available. Neat iso-octane has previously been investigated in the PFR (Lenhert, 2004a). The PFR was operated under the controlled cool down methodology with 8 atm pressure, an equivalence ratio of 1.0, a residence time of 250 ms, and over the temperature range of 600 to 765 K. The start of the NTC regime was at 665 K with a peak production of approximately 250 ppm CO and 350 ppm CO₂. The major alkene intermediate that was produced during the oxidation process was 2-methyl-1-propene with a peak production of approximately 175 ppm at 670 K. The major aldehyde identified was

formaldehyde with a peak production of approximately 300 ppm at 670 K. A comprehensive review of alkanes used in jet fuel surrogates was reported (Colket et al., 2007).

2.4 Chemistry of n-Decane

n-Decane is a moderate molecular weight linear alkane that may be used as potential surrogate component for Fischer-Tropsch jet fuel. There are relatively few detailed and validated n-decane reaction mechanisms available which include low temperature reaction pathways at elevated pressures. Several studies have focused on modeling n-decane at the low and intermediate temperature regimes. The development of models at lower temperatures requires additional speciation data to determine the dominant reaction paths.

The oxidation of n-decane at low and intermediate temperatures has been experimentally studied by Dagaut et al., (1994) for stoichiometric n-decane in a jet-stirred reactor at 550-1150 K temperature, 10 atm pressure, and 1 s residence time. A model was developed and focused on the production of cyclic ethers at low temperatures. However, the model's ability to transition from low and intermediate to high temperatures was poor and it was concluded that increasing the importance of peroxy radicals was necessary.

A reaction mechanism with 600 reactions and 67 species based on flat-flame burner experiments predicted the NTC behavior seen in the transition from low and intermediate to high temperatures (Bikas et al., 2001). However, when this model was applied to recent data, it did not predict the measurements very well (Zhao et al.,

2005). Overall, these n-decane models perform reasonably well at high temperatures but do not capture the relevant low and intermediate temperature behavior.

Several researchers have focused on developing models for a wide range of n-alkanes including n-decane. A unified model for high molecular weight alkanes, including n-decane, simulated autoignition delay times (Buda et al., 2005).

Validation experiments were conducted in shock tubes and rapid compression machines. Ranzi et al. (2005) conducted a wide range kinetic modeling study of the partial oxidation and combustion of n-alkanes up to C_{16} . The mechanism included lumping techniques to simplify the model without losing chemical detail and was experimentally validated with a broad range of experimental conditions. Sirjean et al. (2008) generated a detailed chemical kinetic reaction model used for modeling pyrolysis and oxidation of n-alkanes up to C_{12} at high temperatures. The model contains 1459 reactions and 194 species. Validation tests were conducted with laminar flame speed measurements, ignition delay times and species profiles with shock tubes, and species concentration profiles with jet stirred and flow reactors. Westbrook et al. (2009) developed comprehensive detailed chemical kinetic reaction mechanisms for combustion of n-alkanes from C_8 - C_{16} . These mechanisms include both high temperature and low temperature reaction pathways. The n-decane reaction mechanism contains 3878 reactions with 940 species. Validation of the mechanism includes experimental data from shock tube ignition, rapid compression machine ignition, jet-stirred reactors, flow reactors, and laminar flame experiments.

2.5 Chemistry of Cycloalkanes

Research exploring the branching pathways of cycloparaffins at low and intermediate temperatures is scarce. Research on cyclopentane and cyclohexane in shock tubes was performed over equivalence ratios 0.5-2, a temperature range 1230-1800 K, and a pressure range of 7.3-9.5 atm (Sirjean et al., 2007). The onset of ignition was detected with excited OH radical emission using a photomultiplier tube. The shock tube experiments showed that cyclopentane is much less reactive than cyclohexane, with the autoignition delay times for cyclopentane being 10 times longer than cyclohexane. Cyclohexane was much more reactive due to the formation of cyclohexyl radicals, in which 51 % of the cyclohexyl radicals react to 1-hexen-6-yl, and 45 % react to form cyclohexene and hydrogen radicals. Whereas, cyclopentane reacts to form cyclopentyl radicals in which 80 % of cyclopentyl radicals react to 1-penten-5-yl radicals and 12 % react to form cyclopentene and hydrogen radicals. The formation of the additional hydrogen radicals in cyclohexane reactions promotes a branching step with oxygen molecules to produce oxygen atoms and hydroxyl radicals which support chain branching pathways. The additional chain branching pathways are ultimately responsible for the increase in reactivity of cyclohexane compared to cyclopentane.

In our facilities, the low temperature oxidation (600-800 K) of methylcyclohexane was studied (Lenhert, 2004b). Intermediate species identification and quantification showed that methylcyclohexane (MCH) undergoes a dehydrogenation process at preignition conditions in contrast to previous investigators who proposed the removal of the methyl group. Hydrogen abstraction from the ring is followed by molecular oxygen addition and internal isomerization,

therefore, the ring tends to remain closed. Alkenes are then produced from the decomposition of the molecule.

Another study (Natelson et al., 2007), comparing the reactivity of methylcyclohexane and n-butylcyclohexane (nBCH), found that the MCH was unreactive alone whereas nBCH was reactive. Recently, a low temperature oxidation mechanism for methylcyclohexane was developed based on autoignition experiments in a rapid compression machine (Pitz et al., 2007). The ignition delay times, indicated that the oxidation of methylcyclohexane proceeds through the NTC regime similar to straight chain alkanes. Another study showed that aromatic components were produced from cycloparaffin flames (McEnally et al., 2005). The comparison of aromatic species production from cycloparaffins and their conjugate alkenes showed similar conversions and it was argued that the aromatics were produced from cyclization of the alkenes and not direct dehydrogenation reactions of the rings. However, it is important to note these flame experiments emphasize the high temperature oxidation pathways.

Research on cyclohexane has also expanded and included low temperature combustion studies. Among the experiments used for model comparison were autoignition studies in a rapid compression machine at temperatures of 600-900 K (Lemaire et al., 2001). A model for the oxidation of cyclohexane was developed for low to high temperatures (Silke et al., 2007). To predict the production of benzene, the model included dehydrogenation of cyclohexane. An additional cyclohexane mechanism has been developed based on density functional theory (Cavallotti et al., 2007).

2.6 Chemistry of Petroleum and Alternative Jet Fuels

This section will discuss petroleum derived JP-8 and two different alternatives to petroleum derived jet fuel. All jet fuel samples were acquired from Wright-Patterson Air Force Base (WPAFB). The first fuel that will be discussed is petroleum JP-8, POSF 3773, manufactured from a conventional refining process. The second fuel that will be discussed is a Fischer-Tropsch jet fuel, POSF 4734, derived from natural gas. The third is coal-derived jet fuel, POSF 4765. Two methods of producing coal-derived jet fuel, direct and indirect liquefaction, will also be discussed. Table 2.1 shows an analysis of fuel properties provided with the samples.

Table 2.1: Comparison of selected properties of the jet fuel samples.

Jet Fuel Sample	Petroleum POSF 3773	F-T natural gas POSF 4734	Coal POSF 4765
Aromatics	15.9% vol	0.0% vol	1.8%
Alkenes	0.7% vol	-	-
Cetane Index	45.8	-	-
API Gravity	44.6	55.3	31.1
Heat of Combustion	43.3 MJ/kg	44.1 MJ/kg	43.0 MJ/kg
Hydrogen Content	13.9% mass	15.3% mass	13.2% mass
Fuel System Icing	0.07% vol	-	0.00% vol
Inhibitor			
Total Sulfur	0.07% mass	0.00% mass	0.00% mass
IBP Distillation	150°C	153°C	181°C
10% Distillation	170°C	169°C	192°C
20% Distillation	176°C	176°C	194°C
50% Distillation	196°C	201°C	204°C
90% Distillation	237°C	249°C	243°C
EP Distillation	256°C	271°C	270°C

2.6.1 Petroleum JP-8

Petroleum JP-8 is derived from the distillate fuel fraction in a conventional refining process and is used as the standard fuel for the United States Air Force aviation power systems. Fractional distillation utilizes crude oil, which contains many hydrocarbons with different molecular masses, to produce various petroleum based products. The distillation process includes heating the crude oil and then directing it to a fractionation column where the oil is separated based on boiling points. Three main fractions are produced from the distillation process: the naphtha fraction (125-160°C boiling range) consists mainly of lighter hydrocarbons which are used as gasoline products; the middle distillate kerosene fraction (160-250°C boiling range) consists of hydrocarbons which are processed to kerosene and jet fuels; and heavy residuum distillate fraction (250-350°C boiling range) which is processed to manufacture diesel and heating oils (Speight, 2006). As with most real fuels, JP-8 contains many different classes of hydrocarbons and a wide variability in chemical composition.

The petroleum JP-8 sample used in this study is designated by WPAFB as POSF 3773. As with other JP-8 fuels, POSF 3773 contains linear and branched alkanes, cyclo-alkanes, aromatics, and alkenes as shown in Table 2.2 (Holley et al., 2007). As mentioned in Section 1.1 and shown in Table 1.1, the specifications for JP-8 are based on general physical properties rather than specific chemical composition which leads to wide variations in chemical composition between different samples of JP-8. Holley et al. (2007) compared POSF 3773 to a “world survey average” blended sample of JP-8, US commercial jet fuel (Jet A), US Navy jet

fuel (JP-5), and Russian jet fuel (TS-1), and POSF 3773 was shown to represent typical composition of the jet fuels. Natelson et al. (2008) showed that POSF 3773 represented typical low temperature reactivity behavior of three JP-8 samples. Therefore, the sample of “typical” petroleum derived JP-8 which is assigned as POSF 3773 was used in this study.

Table 2.2: Composition of petroleum JP-8, POSF 3773.

Class	Composition (%)
Paraffins (n- and i-)	57.2
Cycloparaffins	17.4
Dicycloparaffins	6.1
Tricycloparaffins	0.6
Alkylbenzenes	13.5
Indans/tetralins	3.4
Indenes	<0.2
Napthalene	<0.2
Napthalenes	1.7
Acenaphthenes	<0.2
Acenaphthylenes	<0.2
Tricylic aromatics	<0.2

2.6.2 Fischer-Tropsch Jet Fuel

The chemical composition of Fischer-Tropsch jet fuel differs greatly from that of petroleum-derived JP-8. Our sample, designated by Wright-Patterson Air Force Base as POSF 4734, is called S-8 by its manufacturer, Syntroleum, and is composed entirely of alkanes as shown in Table 2.3 (Shafer et al., 2006). Syntroleum manufactured this jet fuel using natural gas as the feedstock to produce a syngas of CO₂ and H₂. The syngas is then converted to liquid hydrocarbons by utilizing the Fischer-Tropsch process. Further analysis has been conducted to determine the major

components of S-8 (Bruno et al., 2006). Table 2.4 shows the components identified utilizing gas chromatography/mass spectrometry. Each component had a peak chromatographic area count greater than 1% of the total raw total ion chromatogram (TIC) area. The major species identified, with an area count greater than 2%, were n-dodecane, n-undecane, and 4-methyloctane. Note that the fuel contains a mixture of linear and lightly branched alkanes of C₉ to C₁₅.

Table 2.3: Composition of Fischer-Tropsch Jet Fuel, POSF 4734.

Class	%
Alkanes	99.7
Monocycloalkanes	<0.2
Dicycloalkanes	0.3
Tricycloalkanes	<0.2
Alkylbenzenes	<0.2
Indans/tetralins	<0.2
Naphthalenes	<0.2
Substituted naphthalenes	<0.2

Table 2.4: Components in Fischer-Tropsch Jet Fuel, POSF 4734.

Components > 1%	
2,5-dimethylheptane	2,6-dimethyldecane
4-methyloctane	4-ethyldecane
3-methyloctane	5-methylundecane
n-nonane	2-methylundecane
4-methylnonane	3-methylundecane
3-methylnonane	n-dodecane
n-decane	x,y-dimethylundecane
2,5-dimethylnonane	2,4-dimethylundecane
5-methyldecane	2-methyldodecane
4-methyldecane	n-tridecane
2-methyldecane	n-tetradecane
3-methyldecane	n-pentadecane
n-undecane	

As shown in Table 2.4, real hydrocarbon fuels contain many components (this list of 25 species only contains species with molar percentages greater than 1%) and chemical compositions that vary. Therefore, a high priority objective of several research groups is to develop surrogates for the real fuels to aid in experimental and modeling studies. The surrogates typically contain 2-10 components whose combustion characteristics are known. The selection of surrogate components and

their proportions are based on matching physical and chemical targets for the particular application. A key target for reciprocating engine applications is auto-ignition which depends on low and intermediate temperature reactivity.

2.6.3 Coal Derived Jet Fuel

Coal-derived jet fuel also has a chemical composition different from conventional petroleum derived jet fuel. The fuel can be produced from methods such as direct and indirect liquefaction. Direct liquefaction consists of reducing the coal to a solvent at elevated temperature and pressure. The elevated temperature promotes the cracking of the carbon-carbon bonds of the coal. The elevated pressure preserves the solvent in liquid phase. Hydrogen gas is bubbled through the solvent, and aided by a catalyst which increases the rate of reactions, hydrogenates the solvent to produce liquid hydrocarbon fuel (Williams et al., 2003; Lumpkin, 1988). The liquid hydrocarbon fuel produced contains high concentrations of cyclic compounds with relatively low concentrations of linear alkanes (Balster et al., 2008). The coal-derived jet fuel used in this study, designated as JP-900 POSF 4765, was manufactured using such a direct liquefaction technique and contains mostly cycloparaffins, as shown in Table 2.5. Another method used to produce jet fuel from coal is the indirect liquefaction technique. The indirect liquefaction method involves gasifying the coal to produce H_2 and CO, also known as syngas. The syngas can then be made into liquid hydrocarbons by the Fischer-Tropsch process (Larson et al., 2003).

Table 2.5: Composition of coal derived jet fuel, POSF 4765.

Class	Composition (%)
Paraffins	<1.0
Cycloparaffins	97.3
Alkylbenzenes	<1.0
Indans/tetralins	1.6
Napthalene/napthalenes	<1.0

2.6.4 Blends of Alternative Jet Fuels

There are issues with operating power systems with 100 % alternative fuels. The lack of aromatics in the alternative fuels can be detrimental to the seals in current power systems. Fischer-Tropsch jet fuel causes the seals to shrink and fail (DeWitt et al., 2008). However, alternative fuels can be mixed with petroleum-derived JP-8. A 50/50 blend of Fischer-Tropsch jet fuel with petroleum JP-8 satisfies the military's requirements for a fuel to be identified as JP-8. If the concentration of Fischer-Tropsch jet fuel is increased to more than 50 %, the minimum specific gravity is outside the acceptable range.

Corporan et al. (2007) developed and studied mixtures of Fischer-Tropsch with petroleum JP-8 in engine experiments. A T63 turboshaft engine and an atmospheric swirl-stabilized research combustor were used to investigate emission characteristics of traditional petroleum derived JP-8, natural gas derived Fischer-Tropsch jet fuel, and blends of the petroleum and Fischer-Tropsch jet fuel. A wide range of temperatures and equivalence ratios were tested. The analysis revealed a notable reduction in particulate matter with the use of the Fischer-Tropsch/petroleum JP-8 blend on both the turboshaft engine and research combustor. Also, with the use of the Fischer-Tropsch/petroleum JP-8 blend, reductions of over 50 % of particulate

mass, and a reduction in smoke number with the engine tests were observed.

Polycyclic aromatic hydrocarbon (PAH) species were not detected in the soot samples collected with blends of more than 50 % Fischer-Tropsch jet fuel.

Additional benefits include a reduction in SO_x levels, as a result of the sulfur free characteristic of synthetic jet fuel addition to the blend, and no measurable fuel usage difference or negative engine performance were observed. (Corporan et al., 2007).

2.7 Closure

This chapter presented an overview and review of ongoing combustion research and modeling efforts related to n-alkanes, cycloalkanes, petroleum JP-8, and alternative jet fuels in the low and intermediate temperature regimes. The oxidation of hydrocarbons in these temperature regimes is not a straight forward combustion process that only produces CO_2 and H_2O . Many intermediate reactions are occurring. As the molecular weight of the hydrocarbon increases, more intermediate species are produced. To predict the combustion behavior of these hydrocarbons, chemical kinetic models are developed. The models can predict a wide range of combustion characteristics such as laminar flame speeds, ignition delay times, and species concentration profiles.

Petroleum JP-8, Fischer-Tropsch natural gas derived jet fuel, and coal derived jet fuel were also discussed. Petroleum derived JP-8 is derived from the distillate fuel fraction in a conventional crude oil refining process whereas the natural gas and coal derived jet fuels are manufactured from non-petroleum based feedstocks. Producing liquid jet fuels from different sources and processing methods leads to differences in

chemical composition. To meet the current specifications of a fuel to be identified as JP-8, blends of alternative fuel with petroleum derived JP-8 need to be utilized.

CHAPTER 3: EXPERIMENTAL FACILITY

3.1 Introduction

The oxidation of alternative fuels for petroleum JP-8 was studied with the aid of the pressurized flow reactor (PFR), single cylinder research engine facility, and the analytical chemistry lab, consisting of a gas chromatograph (GC) with a flame ionization detector (FID) coupled to a mass spectrometer (MS). Section 3.2 describes the pressurized flow reactor facility, which was used extensively throughout the course of this research for preignition studies, and Section 3.3 discusses the single cylinder research engine facility, which was used to investigate autoignition reactivity. The PFR was operated using a controlled cool down methodology, which consists of heating the flow reactor and allowing it to cool at a controlled rate, as described in Section 3.4. Samples extracted from the PFR, throughout the low and intermediate temperature regime, were analyzed online utilizing an Ultramat 22P CO/CO₂ analyzer, as described in Section 3.5. Offline analysis was achieved by storing samples in a heated sample storage system, described in Section 3.6, for transportation of the collected samples from the pressurized flow reactor to the analytical facility. Section 3.7 describes the analytical chemistry lab which was used to analyze the collected species. Section 3.8 contains subsections that describe the upgrades and replacement of parts for the pressurized flow reactor facility and analytical equipment. A brief closure to this chapter is presented in Section 3.9.

3.2 Pressurized Flow Reactor Facility

The PFR, shown in Figure 3.1, is designed to study combustion chemistry with relative isolation from fluid mechanics and temperature gradients (Koert et al., 1992). The PFR can safely reach temperatures up to 850 K. The PFR is designed to withstand pressures up to 20 atm, however, during this study, 8 atm was the maximum experimental pressure. The key operational feature of the PFR is a quartz reactor tube within a pressure vessel. The inside of the reactor tube and the annulus between it and the pressure vessel are maintained at the same pressure, which is controlled by a pressure regulating valve. Synthetic air, composed of nitrogen (purity = 99.9%) and oxygen (purity = 99.994%), is heated to the reaction temperature. A high pressure syringe pump (Isco 500D) injects the liquid fuel into the centerline of a heated nitrogen stream to ensure vaporization. The synthetic air and the prevaporized fuel/nitrogen mixture are mixed in an opposed jet annular mixing nozzle at the entrance of the quartz reactor tube. A water-cooled, borosilicate glass-lined stainless steel probe extracts samples from the centerline of the quartz reactor tube and quenches the chemical reactions. A type-K thermocouple is also integrated into the probe assembly to measure the sample temperature. To limit temperature rise due to heat release, nitrogen is added to the fuel. The nitrogen dilution of the fuel is defined by,

$$\% \text{ Dilution} = \frac{Q_{\text{Nitrogen}} + Q_{\text{Oxygen}} - 4.76Q_{\text{Oxygen}}}{Q_{\text{Nitrogen}} + Q_{\text{Oxygen}}}, \quad (1)$$

where Q_{Nitrogen} and Q_{Oxygen} are the total volumetric flow rate of nitrogen and oxygen.

Temperature rise is monitored by comparing inlet and sample temperatures; average temperature rise was 25 K, with maximum values of approximately 35 K at peak reactivity. To minimize heat loss and to maintain isothermal conditions of the PFR, the walls of the pressure vessel are insulated and the reactor is heated with eight 800 W independently controlled bead heaters.

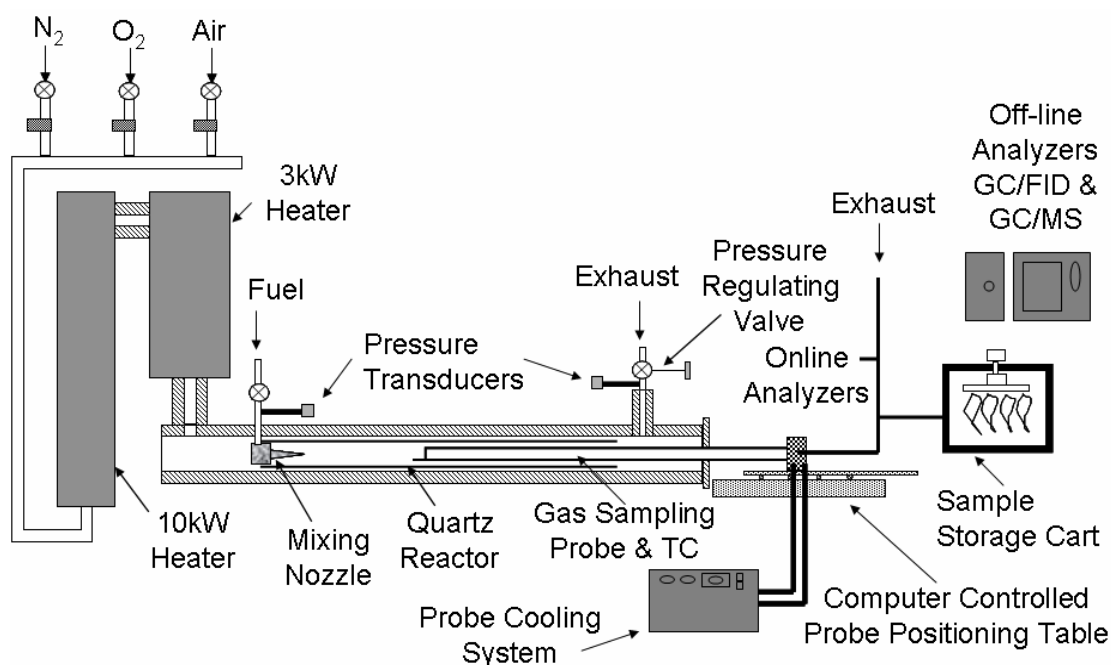


Figure 3.1: Schematic of the PFR identifying key components.

3.3 Single Cylinder Research Engine Facility

Engine experiments were conducted with a modified single cylinder, variable compression ratio Cooperative Fuel Research (CFR) engine coupled to a dynamometer, Fig. 3.2. The engine has a moveable cylinder head which allows for variable compression ratios. For this study, the compression ratio was held constant at 16:1. The bore is 8.25 cm, the stroke is 11.43 cm, and the displacement is 611 cm³.

For each experiment, the inlet manifold pressure was 0.1 MPa, the engine speed was 750 RPM, and the equivalence ratio was 0.478. Experiments were run at an inlet temperature of 480 K, above the boiling points of the fuels in this study and thus eliminating concerns about fuel condensation. The fuels were injected into the air stream well upstream of the heated inlet manifold and the fuel/air mixture passed through turbulators to ensure mixing and complete vaporization. The engine was operated under a premixed combustion ignition (PCI) mode controlled by the autoignition process. Figure 3.2 shows the main components of the single cylinder research engine facility: heated intake manifold, inlet and exhaust thermocouples, fast sampling valve for offline analysis, gas analyzers, and data control system. Table 3.1 lists selected CFR engine specifications.

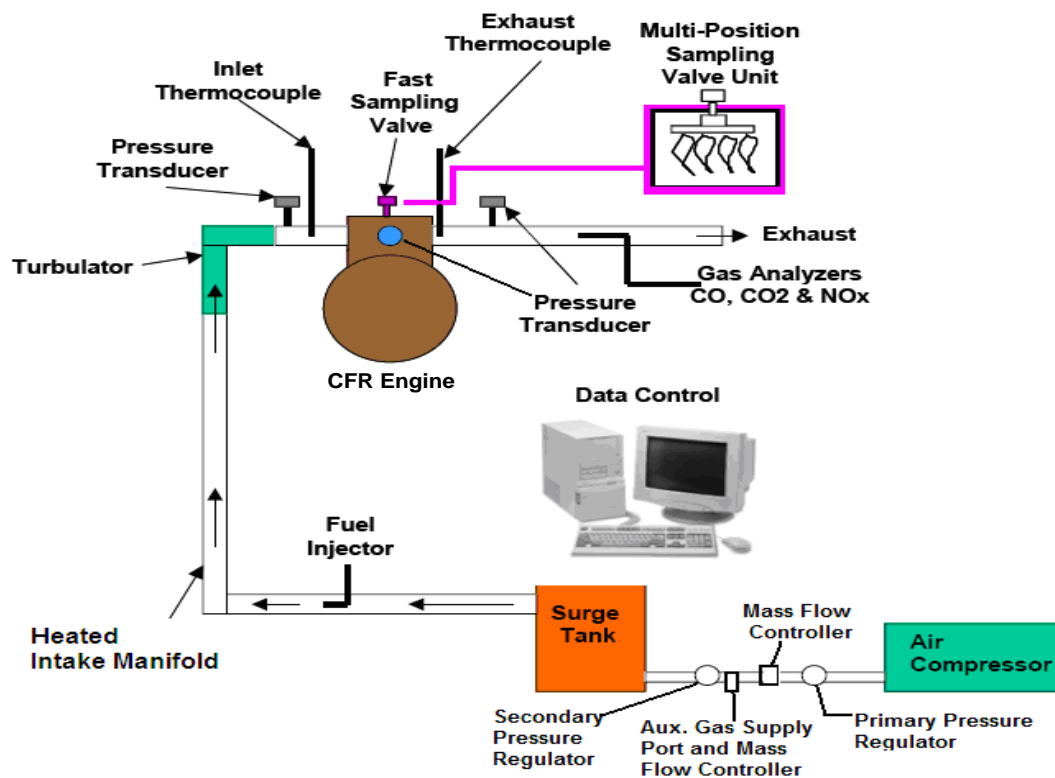


Figure 3.2: Schematic of the single cylinder research engine facility with key components identified.

Table 3.1: Selected research engine specifications.

Stroke	114.3 mm
Displacement	611.6 cm ³
Compression Ratio	4:1-18:1
Intake Valve Opens	10° before TDC
Intake Valve Closes	34° after BDC
Exhaust Valve Opens	40° before BDC
Exhaust Valve Closes	15° after TDC

3.4 Controlled Cool Down

For the PFR experiments, a methodology known as controlled cool down (CCD) was followed, in which pressure and residence time were held constant while the temperature varied. A controlled cool down experiment is performed by heating the reactor with the 10 kW and 3 kW air circulation heaters, and the eight 800 W bead heaters to the desired temperature for the study. Once the temperature is reached, the heaters are turned off, and the reactor cools at a rate of 2-5 K/min. The probe position is adjusted to maintain a constant residence time as the temperature changes. The extracted gas sample flowed through a heated (493 K) sample line to a nondispersive infrared (NDIR) analyzer for CO and CO₂ measurements. Experimental error is ± 25 ppm. At these temperatures, CO has been shown to be a good indicator of reactivity; CO does not oxidize to CO₂ significantly and separate pathways produce the species (Wilk et al., 1989). The start of NTC occurs at the temperature of maximum CO production. Table 3.2 shows the typical operating conditions for an experiment in the PFR utilizing the CCD methodology.

Table 3.2: Typical experimental operating conditions of the PFR.

Temperature Range	600-800 K
Pressure	8 atm
Equivalence Ratio	0.30
Residence Time	120 ms
N ₂ Dilution in Fuel	80.0%

3.5 Siemens Ultramat 22P CO and CO₂ Detector

During the controlled cool down experiments, the Siemens Ultramat 22P (7MB1123-1FF13-2AA1) analyzer was used online with the PFR to monitor CO and

CO₂. Once the sample is extracted from the PFR, the sample goes through the heated sample transfer line to the sample storage cart. The sample storage cart is connected to the Ultramat 22P with unheated 1/8" stainless steel tubing. The tubing is unheated to allow high boiling point species to condense before entering and damaging the analyzer. A sample flow of 3 l/min was maintained by a stainless steel needle valve. The sample is then analyzed by the Ultramat 22P for CO and CO₂ concentrations. The exhaust from the Ultramat 22P is plumbed directly into the fume hood that is located near the PFR facility. The CO and CO₂ channels were checked periodically with calibration standards to test reproducibility. The reproducibility was between 3-5% of each calibration standard tested. Experimental accuracy of the detector was \pm 25 ppm.

However, during the course of this research, the Ultramat 22P was experiencing some operational problems. There were several occasions where the analyzer would randomly go to 5535 ppm for the CO measurement. When this would happen, the analyzer would be turned off and then turned back on. Following the restart, the analyzer would function properly. It is recommended for future work that the Ultramat 22P be replaced.

3.6 Sample Storage System

The sample storage system consists of a cart mounted Cole-Parmer heated oven which contains a 16 position electronically controlled multi-position valve with 16 10-mL stainless steel sample storage loops. The oven was maintained at a temperature of 190 °C during experiments and placed in the sample train before the

CO/CO₂ detector. Fifteen sample loops were used for sample storage purposes and the other loop was needed to switch from sample collection to sample analysis mode.

When connected to the PFR via the heated transfer line, which was set to 190 °C, sample gas continuously flowed through a 10 mL sample loop, then to the CO/CO₂ detector, then exited to an exhaust vent. The on/off valve A is in the open position for sample to flow to the CO/CO₂ detector during sample collection mode. The sample was continuously monitored for CO and CO₂. When a sample needs to be collected, the step button on the electronic actuator was pressed and the multi-position valve switched to the next 10 mL sample storage loop. Each press of the actuator positioned the sample line to an empty loop in which a sample was collected and stored. The remote electronic actuator displayed the position number of the sample loop being filled. This process was continued until 15 samples were collected in 15 different sample storage loops. Figure 3.3 is a schematic of the sample collection system in sample collection mode.

Once all of the sample storage loops were filled, the on/off valve A on the sample storage cart was closed. The storage cart was then disconnected from the flow reactor and the end of the storage cart that was connected to the flow reactor was sealed. The storage cart was then disconnected from the CO/CO₂ detector. Once disconnected, the sample storage cart was moved to the analytical lab to be connected to the GC/MS/FID system. Once connected to the GC/MS/FID system, the on/off valve 1 is opened and a vacuum line from the GC valve oven evacuates the transfer line. After the transfer line is evacuated, the vacuum is turned off. To remove a sample from the sample storage cart, the step button on the remote electronic actuator

is pressed, and a 10 mL sample loop opens to allow the sample gas to exit through the opened on/off valve A. The sample then enters the GC valve oven. Figure 3.4 is a schematic of the sample collection system in sample analysis mode. Details of the injection process and the GC valve oven are presented in Section 3.7.

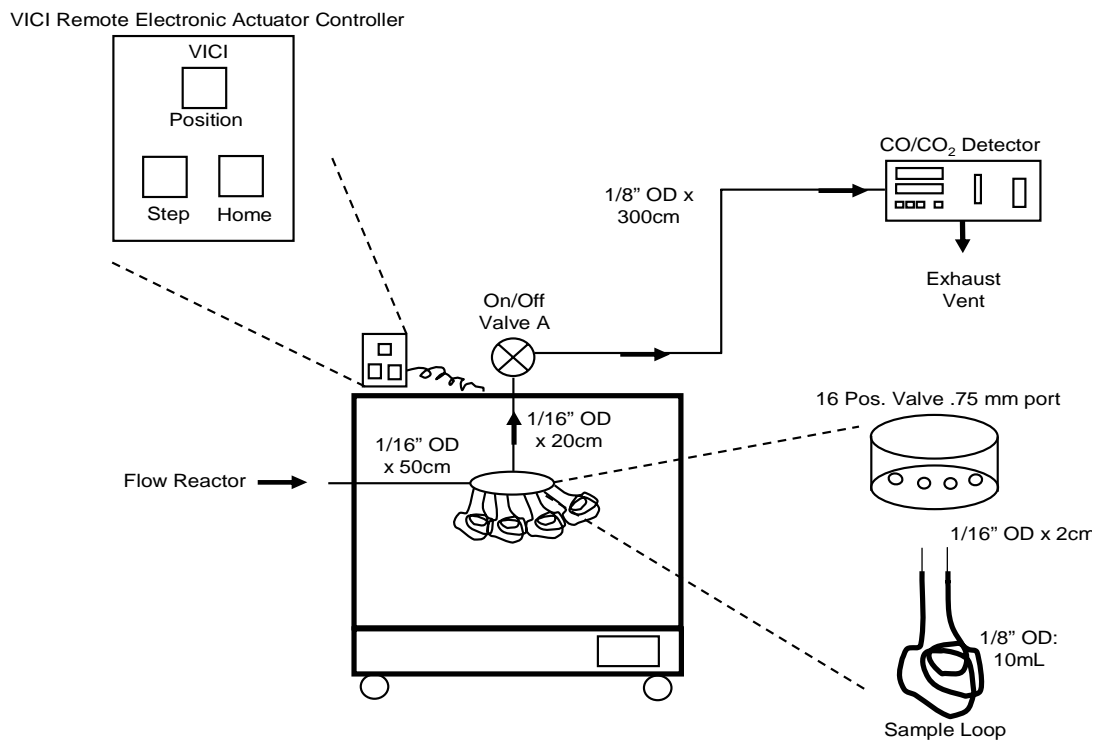


Figure 3.3: Schematic of the heated sample storage system cart in sample collection mode.

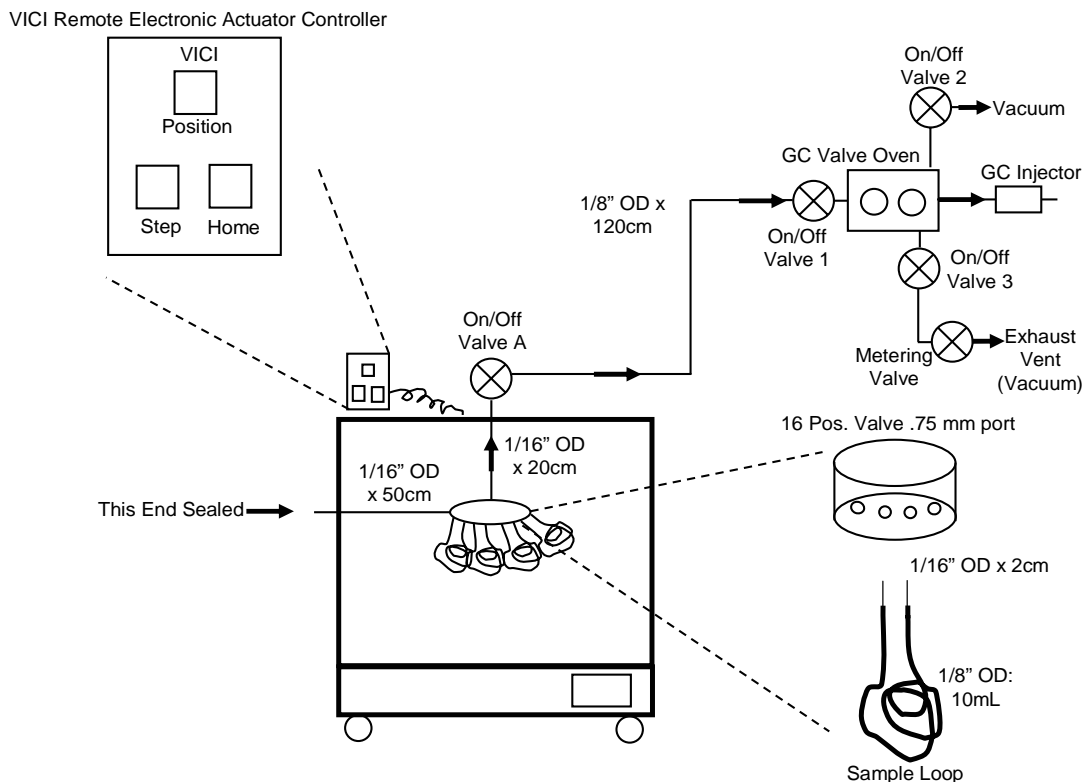


Figure 3.4: Schematic of the heated sample storage system cart in sample analysis mode.

3.7 Trace GC Gas Chromatograph and Trace DSQ Mass Spectrometer

The stable intermediates that were stored in the sample storage cart were analyzed offline using a Thermo Finnigan TraceGC gas chromatograph with flame ionization detection (GC/FID) and the gas chromatograph coupled to the Thermo Finnigan TraceDSQ mass spectrometer (GC/MS). For separation of the species, a Supelco Petrocol DH column (100 m length, 0.5 μm film thickness, 0.25 mm OD) was used. To provide sufficient separation of lighter hydrocarbons, the initial column oven temperature was sub-ambient. Identification was determined both by analyzing the retention time and the mass spectrum of the sample gases. The unknown mass spectra were then compared to the NIST Version 2.0 spectrum database for

identification purposes. Figure 3.5 shows a typical programmed GC temperature profile. Table 3.3 shows typical MS operating parameters.

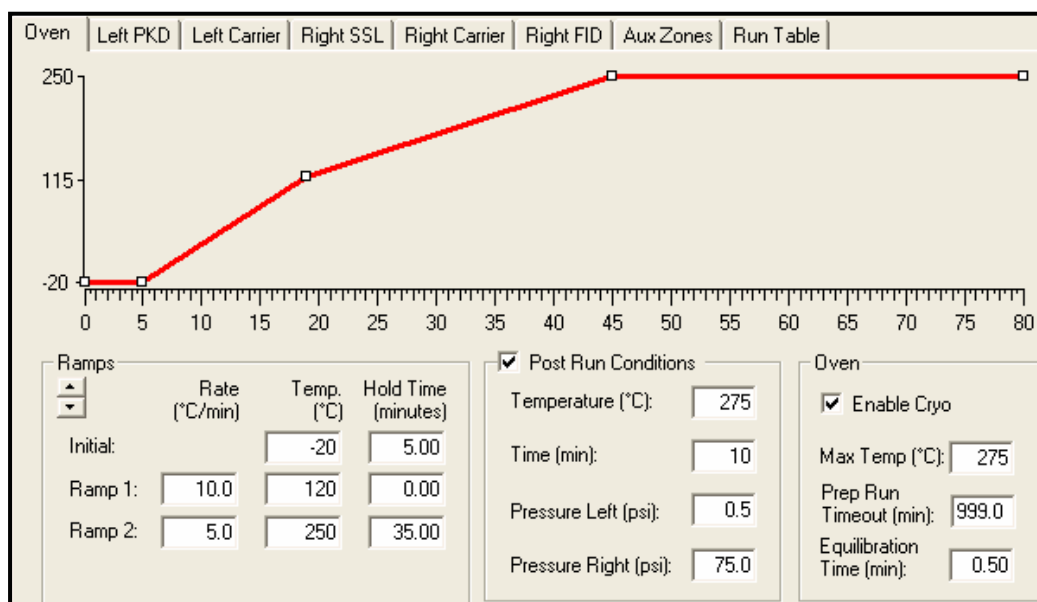


Figure 3.5: Programmed GC temperature profile.

Table 3.3: MS operating parameters.

Parameter	Set Point
Ion Source Temperature	200 °C
Scan Range	10-250 amu/z
Scan Rate	500 amu/sec
Multiplier Voltage	1812 V
Ionization Mode	Electron
Electron Energy	-70 eV
Emission Current	100 μ A
Chromatographic Filter	4 sec

The samples from the sample cart are transferred to the GC/MS/FID system via a heated 1/8" stainless steel transfer line. The line is heated with a bead heater controlled with a Variac to 190 °C. The line is connected to the heated GC valve

oven. The oven contains a 4-port and 6-port multi-position electronically controlled valve. The 4-port and 6-port multi-position valves are necessary to properly control the supply vacuum, sample injection, and helium carrier gas. The 6-port valve allows switching the column flow through the 1 mL sample loop to begin the analysis. The temperature of the GC valve oven is controlled with the key pad on the GC and was set to 190 °C. The 4-port and 6-port multi-position valves have on/off positions that are controlled by the GC, either the key pad or the software. Figures 3.6-3.10 shows the programmed GC valve sequence necessary for analyzing a sample. Figure 3.6 shows both the 4-port and 6-port multi-position valves in the off position. Prior to injecting a gas phase sample into the GC, the 1 mL sample loop is evacuated to ensure no contamination from the previous sample is present. With the 4-port and 6-port multi-position valves in these positions, on/off valve 1 closed, on/off valve 2 open, and the metering valve open, a vacuum is created inside the 1 mL sample storage loop. The pressure inside the 1 mL sample loop can reach 1 Torr with the aid of the vacuum pump. Once vacuum is achieved, the step button on the remote electronic actuator controller is pressed and on/off valve 1 is opened to allow sample to flush out the loop. After a few seconds of flushing, on/off valve 3, the metering valve, and on/off valve 1, are closed. With these valves closed, the sample is stored in the 1 mL sample loop ready to be injected. However, it is necessary to always inject at the same pressure and temperature to make certain the total mass of the sample is constant. For this study, the sample pressure was set at 570 Torr. To achieve this injection pressure, it is necessary to open on/off valve 3 and the metering valve to vent any excess pressure. The pressure is monitored by a Setra pressure

transducer. This position is labeled prep run and is shown in Fig. 3.6. The sample injection position is shown in Figure 3.7. The 6-port valve is turned to the on position, at 0.00 minutes, and the sample is transferred to the injector and GC oven by the helium carrier gas. At 3.00 minutes into the GC program, both the 4-port and 6-port multi-position valves are turned to the off position as shown in Fig. 3.8. In this position, helium carrier gas is continuing to flow through the 1 mL sample loop to transfer the sample through the column of the GC. At 4.00 minutes, the 4-port valve is turned to the on position and the on/off valve 2 is opened. This position allows evacuation of the 1 mL sample loop as shown in Fig. 3.9. Figure 3.10 shows the positions of the 4-port and 6-port multi-position valves at the end of a GC programmed run. Both the 4-port and 6-port multi-position valves return to the off position and are ready to accept the next injection.

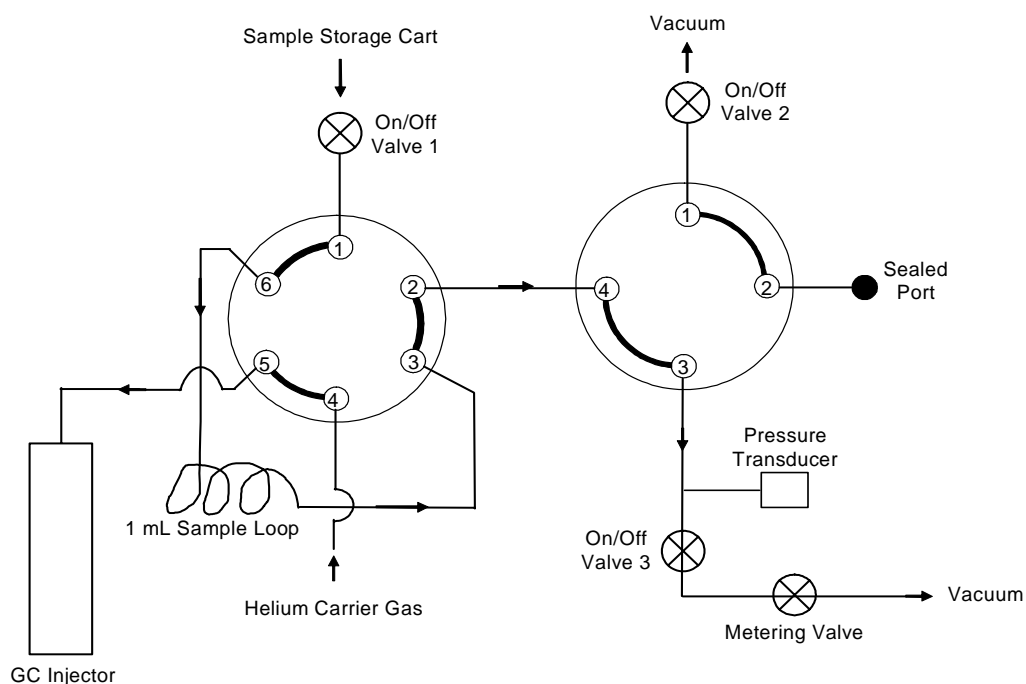


Figure 3.6: 4-port and 6-port valves in the off position, prep run.

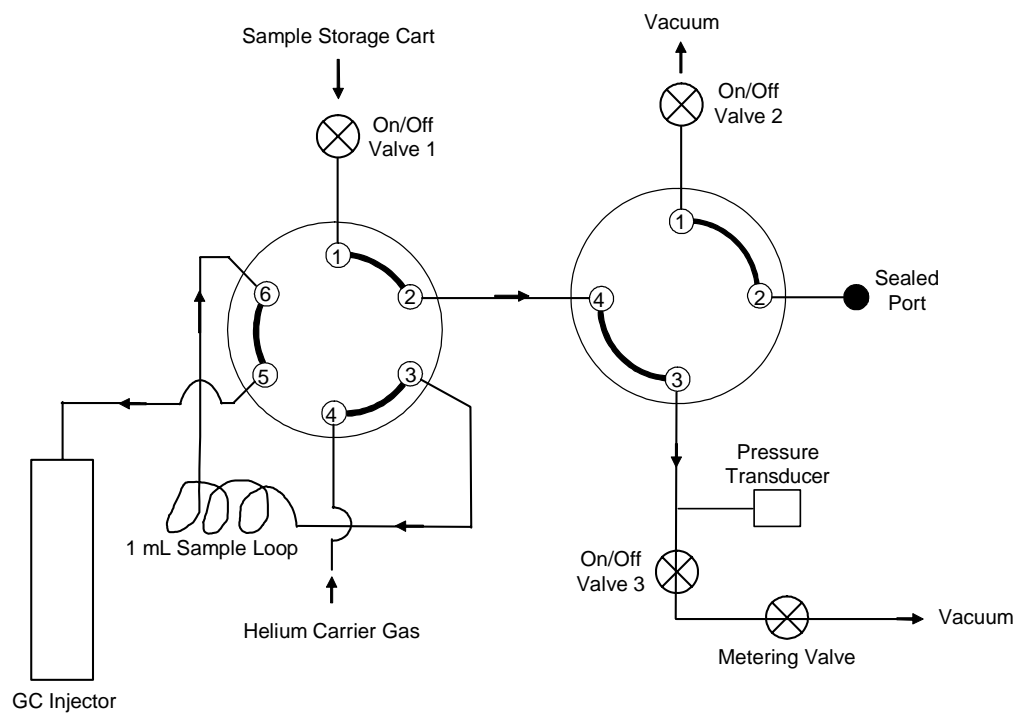


Figure 3.7: GC valves in the sample injection position with the 6-port valve on and the 4-port valve off.

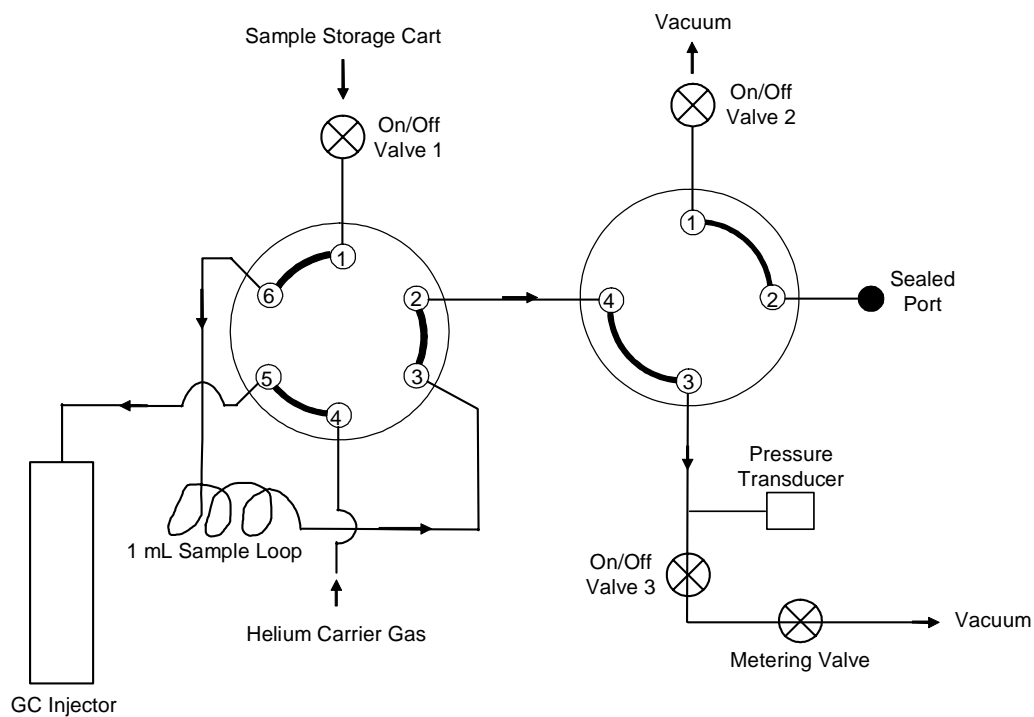


Figure 3.8: 4-port and 6-port valves off in the sample injection position.

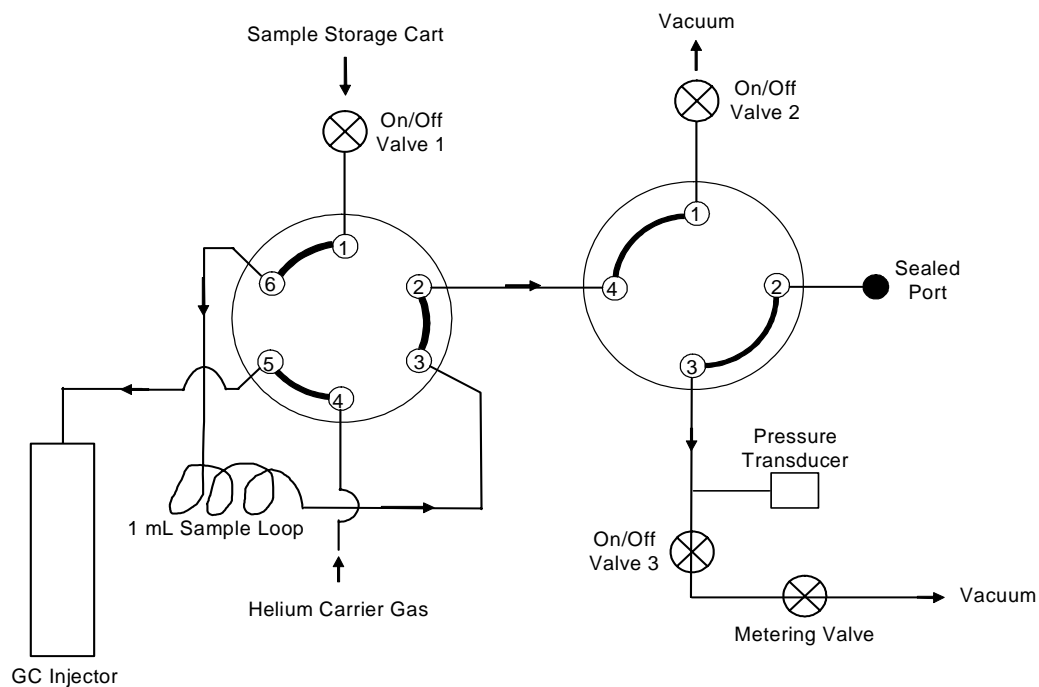


Figure 3.9: Position of GC valves at 4.00 minutes with the 6-port valve off and the 4-port valve on.

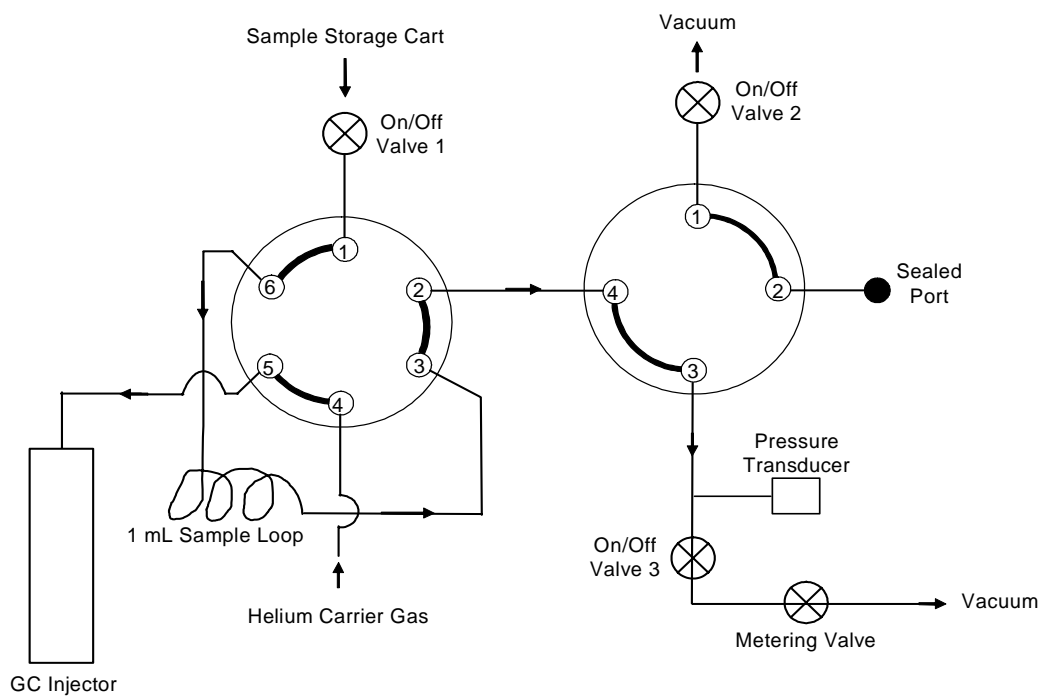


Figure 3.10: Off position of the 4-port and 6-port valves at end of program.

3.8 Pressurized Flow Reactor Facility Upgrades and Maintenance

Throughout the course of this research, several facility upgrades and maintenance procedures were performed. This included updating the sample probe construction and welding procedure, replacing the PFR inlet bead heater controller, replacing the Chromalox 10 kW air circulation heater and controllers, cleaning the 3 kW heater electrical connectors and replacing the insulators, and replacing miscellaneous mass spectrometer and gas chromatograph parts. The following sections provide detailed descriptions of the upgrades.

3.8.1 Sample Probe Construction and Laser Welding

In the past, the technique for manufacturing the sample probe consisted of TIG welding the 3/8" sample probe shaft to the sample probe tip (Koert, 1990 and Lenhert, 2004b). Silver solder was utilized to attach the 1/16" stainless steel sheathed thermocouple and the 1/8" stainless steel glass lined tube to the sample probe tip. The welding was performed in the Drexel University Machine Shop. Conventional TIG welding was sufficient for welding the 3/8" O.D. tube to the sample probe tip. However, to TIG weld the thermocouple and the glass lined tube to the sample probe tip, extreme care had to be taken so as not to damage the thermocouple or the glass lining of the tube. A recent modification of this technique was to silver solder the thermocouple and the glass lined sample tube to the back side of the sample tip and apply stainless steel putty to the face of the tip (Lenhert, 2004b). There were disadvantages with this technique such as the putty being soluble in water and difficulty in the nature of TIG welding such small parts.

The new technique utilized the original design of the sample probe tip, but all of the welding was performed using a laser welder. Figure 3.11 shows the sample probe tip, sample probe shaft dimensions, and laser welding locations. Using this type of welding system, precise welds could be achieved. Extreme care can be given with the laser, therefore reducing the possibility of damaging either the sheathing of the thermocouple or the glass lining of the sample tube. The sample tip was welded to the 3/8" sample probe shaft in this manner as well. Precision Joining Technologies, Inc. in Miamisburg, Ohio, was the vendor that was chosen to perform the laser welding. After the sample probes were returned from the welder, the welds were inspected to check for deterioration or cracking of the glass lining before completing the assembly.

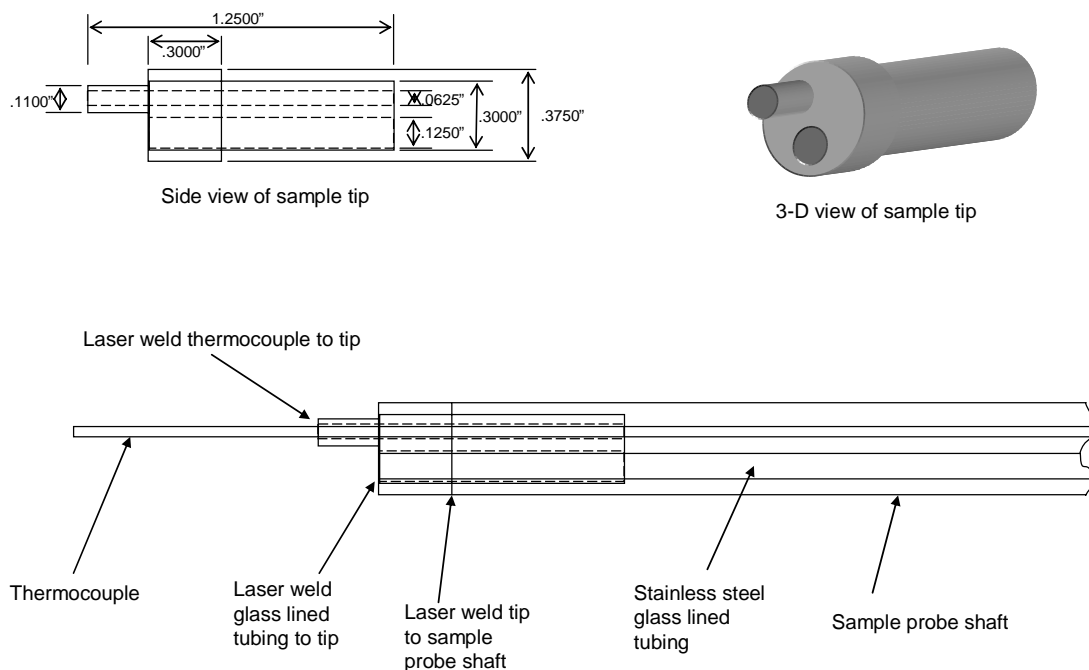


Figure 3.11: Schematic of the sample probe tip showing laser welding locations and dimensions.

3.8.2 Replacement of the Inlet Bead Heater Controller

During a preheat phase of a PFR controlled cool down experiment on November 14, 2006, the inlet bead heater controller failed. While monitoring the temperature of the flow reactor's bead heaters, the inlet temperature dropped rapidly. The rack mounted controller was then removed from the controller assembly. Once removed, the controller was tested and inspected. The transformer inside the controller was at fault and unfortunately the complete controller had to be replaced. Omega model number (Omega, CN1001TC-AI, \$405.00) was used as a replacement controller and was installed on January 29, 2007.

3.8.3 Replacement of the Chromalox 10 kW Air Circulation Heater and Controllers

The main heater for the PFR is the Chromalox 10 kW air circulation heater. The original unit was installed for the facility during the mid 1980's. On January 29, 2007, the preheating phase of a controlled cool-down experiment was being conducted and the 30 Amp, 277/480 Volt, breaker for the main power to the heater tripped. Once the breaker tripped, an initial check of loose wires for any signs of a malfunction revealed no problem. The breaker was reset and the preheat stage of the experiment was continued. During the rest of the preheat stage, the rate at which the temperature increased of the PFR was drastically reduced. Typically, the PFR can reach operating temperature of 800 K in approximately 6-7 hours. During this experiment it took approximately 11 hours to reach operating temperature. The components of the 10 kW heater that were checked included wire connections, the

breaker, and temperature controllers. After thoroughly investigating the cause of the tripped breaker and slow heating time, it was determined that the 10 kW heater assembly was at fault. The wire connection cap was removed from underneath the heater and voltage was found to be present. However, a resistance check of the heater elements showed that 4 of the 6 heater elements were faulty.

The heater is designed so that the subassembly, which includes the heater elements, can be replaced. Initially, just the subassembly (Chromalox, TMISB-6WXX 480V 10KW 1-3P, \$6212.00) was going to be replaced. However, during the disassembly of the 10 kW heater, it was found that the outside of the heater assembly was corroded and holes were present in the housing. The suspected cause of this was due to insulation being wrapped around the housing to prevent heat loss. The insulation was probably trapping moisture which corroded the sheet metal housing of the 10kW heater. It was then determined that the entire 10 kW heater assembly had to be replaced. Chromalox was contacted and a quote was generated for a replacement 10 kW air circulation heater (Chromalox, GCHISB-6WXX 480V 1-3PH 10kw, \$6655.00). The heater was ordered on April 20, 2007. Due to several delays in the manufacturing process, the heater did not arrive until August 7, 2007. The installation of the entire 10 kW heater assembly was completed on August 8, 2007.

There was an ongoing problem with the temperature and overtemperature controllers for the 10 kW heater. The problem was that the overtemperature controller would trip when a fault was not present. With the old heater this was not much of a concern and the overtemperature controller was just reset. However, since the new heater was just installed, it was decided to upgrade both of the controllers as

well. Both the process controller (Chromalox, 2104-R0000, \$390.00) and limit controller (Chromalox, 3101-11000, \$430.00) were purchased from Chromalox. The installation was completed on September 6, 2007.

3.8.4 Cleaning the 3 kW Heater Electrical Connectors

The 3 kW heater assembly utilizes four semi-cylindrical Lindberg heater units wired in parallel with a 115 VAC, 50 Amp power supply. The electrical connections for each heater element are brought outside of the pressure vessel through electrical sealing assemblies. The electrical circuit is made by connecting the eight electrodes as shown in Fig. 3.12, with the individual heater unit connector pairings as shown in Table 3.4. Details of the heater assembly have been discussed (Bhat, 1998). During this research, the efficiency of heating the flow reactor was diminishing. It was suspected that one of the four heater elements may have failed. However, a resistance check of each heater element showed that the elements were fine, as shown in Table 3.4.

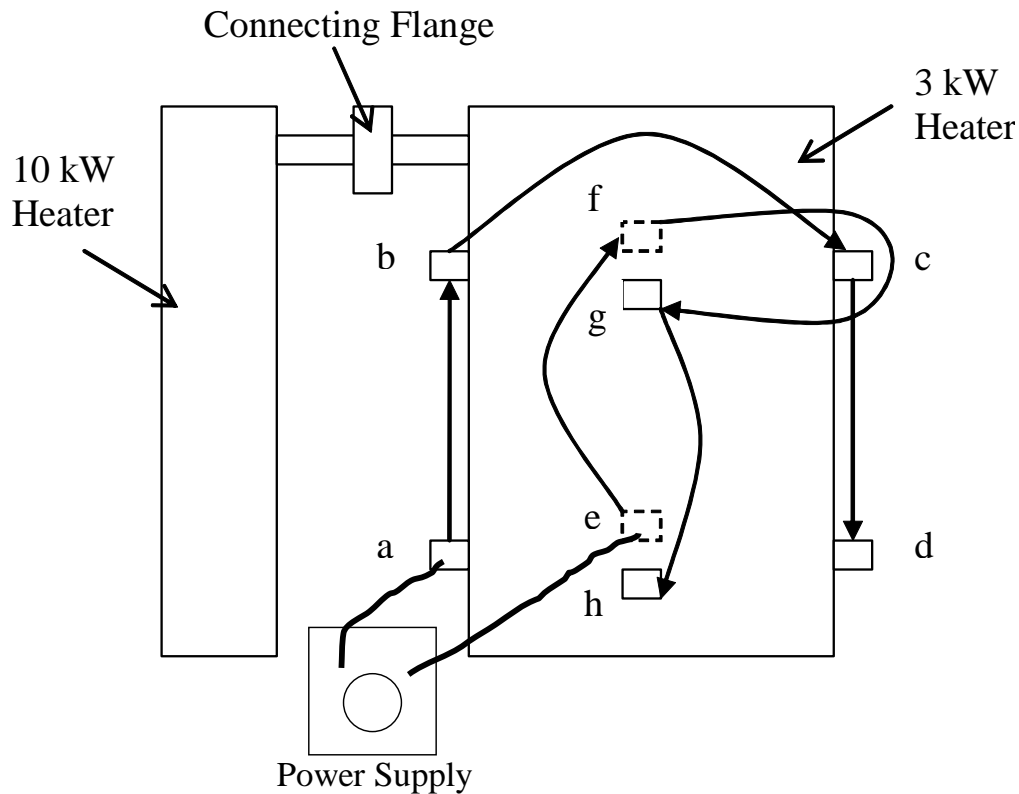


Figure 3.12: Wiring diagram of 3 kW heater elements. Connection e and f are on the backside of the heater.

Table 3.4: Resistance of each heater element.

Connection	Resistance (Ω)
b-g	21.35
a-h	15.96
f-c	21.33
d-e	15.91

The electrical supply wire connects to each of the sealing arrangement's copper electrodes by a copper wire clamp. The connection was found to be poor due to the copper corroding. The sealing arrangements were then removed and cleaned using the glass-bead blaster in Drexel's Machine Shop. Once cleaned, the sealing assemblies were reinstalled and the electrical connections were securely tightened.

3.9 Closure

This chapter described the experimental facility and test procedures, and described modifications to the facility to upgrade capabilities and maintain operation. The experimental facilities included the PFR, used to investigate preignition chemistry, and the single cylinder research engine, used to investigate autoignition behavior. The analytical facility included the GC/MS/FID system and CO/CO₂ detector which were used to detect species produced during the oxidation process. Also, this chapter discussed the necessary replacement of parts and upgrades to the existing research facility.

CHAPTER 4: FISCHER-TROPSCH JET FUEL, PETROLEUM JP-8, AND N-DECANE/ISO-OCTANE OXIDATION

4.1 Introduction

The first component of this study was to examine the low and intermediate temperature reactivity of Fischer-Tropsch jet fuel, compare its reactivity to that of petroleum-derived JP-8, and propose surrogate components for Fischer-Tropsch jet fuel. Section 4.2 presents the results of the investigation and Section 4.3 discusses the implications of this study.

Previous research has shown that it is important to choose a JP-8 with the typical composition and reactivity of all JP-8's (Lenhert, 2004b; Natelson, 2008). As described previously, a sample fitting these requirements was obtained from Wright-Patterson Air Force Base and is identified as JP-8 POSF 3773. A Fischer-Tropsch jet fuel sample, one of only a few samples available at the time of this study, and considered an average sample was also obtained from Wright-Patterson Air Force Base. It is identified by its manufacturer as S-8 and by Wright-Patterson Air Force Base as POSF 4734. Table 4.1 shows properties of these two samples.

A potential Fischer-Tropsch jet fuel surrogate, containing a mixture of n-decane/iso-octane was investigated as well. n-Decane was selected as a component representing the linear alkanes because it is one of the more common high molecular weight alkanes, and it approximates the average chemical formula for JP-8, $C_{11}H_{21}$. iso-Octane (2,2,4-trimethyl-pentane) was selected as a component representing the branched alkanes because it is the most researched moderate molecular weight branched alkane. An initial mixture of 59.4% n-decane / 40.6% iso-octane was

chosen to approximate a cetane number slightly higher than that of JP-8, in anticipation of the higher reactivity of Fischer-Tropsch jet fuel. A linear blending technique was used to determine the cetane number of the mixture, i.e., the cetane number of each component was multiplied by the percentage of the component in the mixture, then adding each of the cetane number percentages together. Table 4.2 shows properties of the mixture and its components. Figure 4.1 shows the chemical structure of the species in the mixture. Section 4.3 presents the closure to Chapter 4.

Table 4.1: Properties of petroleum and Fischer-Tropsch samples.

Property	Petroleum POSF 3773	Fischer-Tropsch POSF 4734
Aromatics	15.9% vol	0.0% vol
Alkenes	0.7% vol	-
Cetane Index	44.8	-
API Gravity	45.8	55.3
Heat of Combustion	43.3 MJ/kg	44.1 MJ/kg
Hydrogen Content	13.9% mass	15.3% mass
Fuel System Icing Inhibitor	0.07% vol	-
Total Sulfur	0.07% mass	0.00% mass
IBP Distillation	150 °C	153 °C
10% Distillation	170 °C	169 °C
20% Distillation	176 °C	176 °C
50% Distillation	196 °C	201 °C
90% Distillation	237 °C	249 °C
EP Distillation	256 °C	271 °C

Table 4.2: Properties of the mixture and components.

Property	n-Decane	iso-Octane	Mixture
Carbon #	10	8	9.1
Hydrogen #	22	18	20.2
Molecular Mass	142.284 g/mol	114.230 g/mol	129.478 g/mol
Cetane #	76	12	50
% Volume	59.4	40.6	-
Density	0.73 g/cm ³	0.69 g/cm ³	0.72 g/cm ³

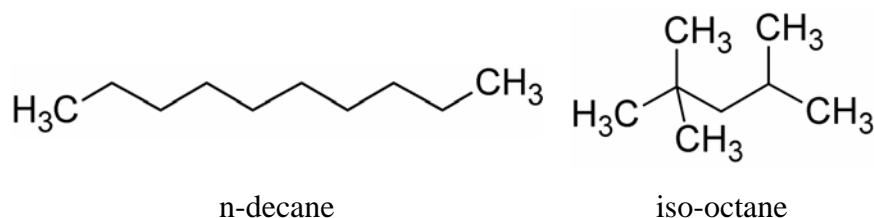


Figure 4.1: Chemical structure of each species in the binary mixture.

4.2 Results and Discussion

This section describes the experimental results of oxidation of the jet fuels and the potential surrogate mixture. For these experiments, the PFR was operated under the controlled cool down methodology, as described in Section 3.4. Subsection 4.2.1 compares the CO and CO₂ production of the petroleum derived JP-8 with the Fischer-Tropsch jet fuel. Subsection 4.2.2 discusses the results of the oxidation of the potential surrogate. Subsection 4.2.3 discusses the results of tuning the surrogate mixture to match the low and intermediate temperature reactivity of Fischer-Tropsch jet fuel.

4.2.1 CO and CO₂ Comparison of Petroleum JP-8 with Fischer-Tropsch Jet Fuel

Figure 4.2 compares the CO production of the petroleum-derived JP-8 and the Fischer-Tropsch jet fuel. The former produced a maximum CO concentration of 600 ppm at 692 K. The latter initiated the NTC region at approximately the same temperature, producing 890 ppm of CO at 690 K. The Fischer-Tropsch jet fuel produced more CO than the petroleum-derived JP-8; however, the reactivity followed similar trends.

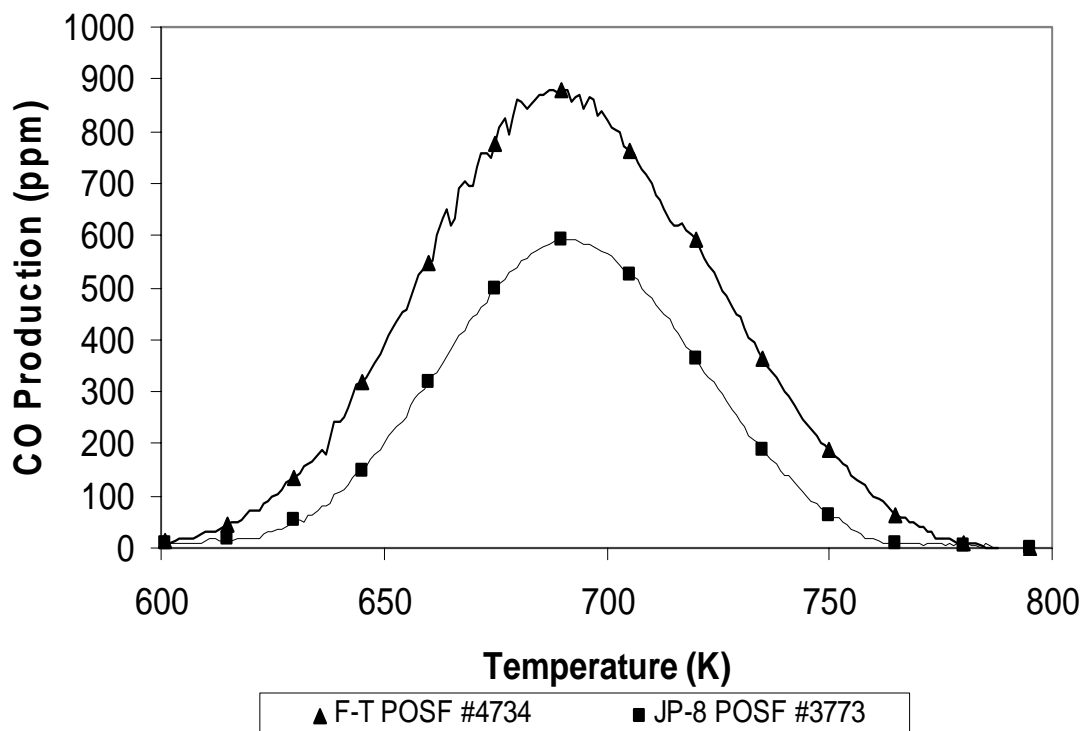


Figure 4.2: CO production of Fischer-Tropsch jet fuel and petroleum JP-8.

The higher reactivity of the Fischer-Tropsch jet fuel is a result of its 100% alkane composition. Because petroleum-derived JP-8 consists of approximately 40% aromatics and cycloalkanes, which have lower cetane numbers and are less reactive than alkanes, the petroleum-derived JP-8 produced significantly less CO. In addition, in the absence of aromatics, which scavenge radicals, higher reactivity of Fischer-Tropsch jet fuel is expected.

Figure 4.3 compares the CO₂ production of the petroleum-derived JP-8 and the Fischer-Tropsch jet fuel. Maximum CO₂ production was approximately 250 ppm for each fuel, and the difference between the fuels is within the 50 ppm experimental error.

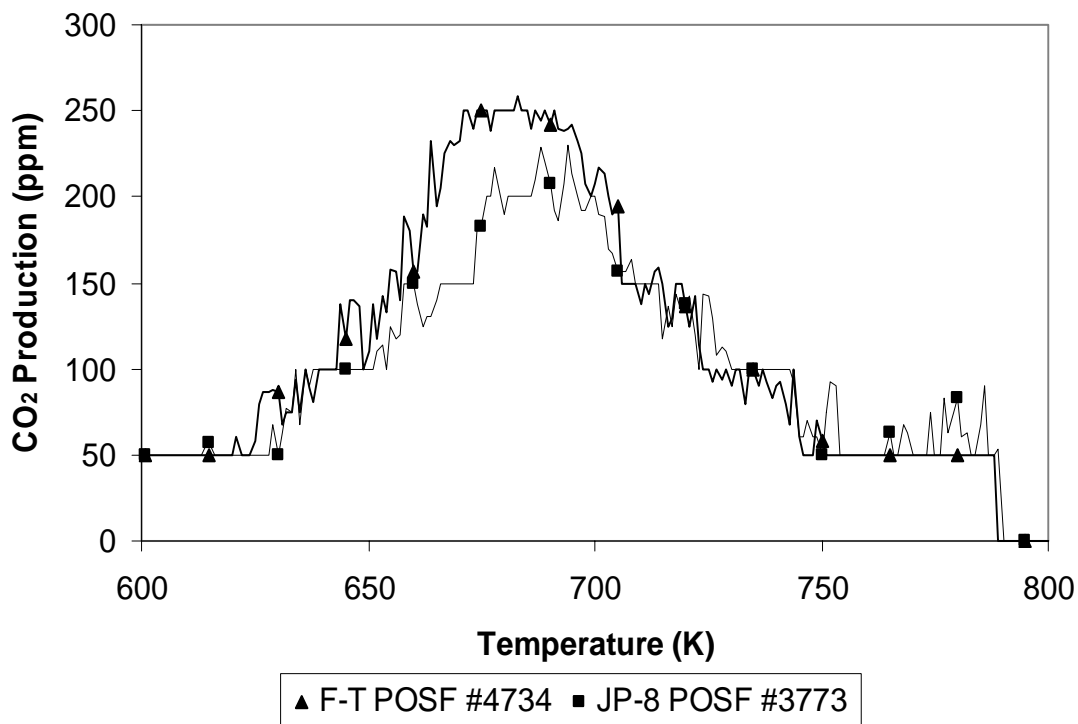


Figure 4.3: CO₂ production of Fischer-Tropsch jet fuel and petroleum JP-8.

4.2.2 Oxidation of the n-Decane/iso-Octane Mixture

A mixture of 59.4% n-decane/40.6% iso-octane by volume was oxidized in the PFR. The CO and CO₂ produced from the mixture are shown in Figure 4.4. The maximum CO concentration was 990 ppm at 694 K.

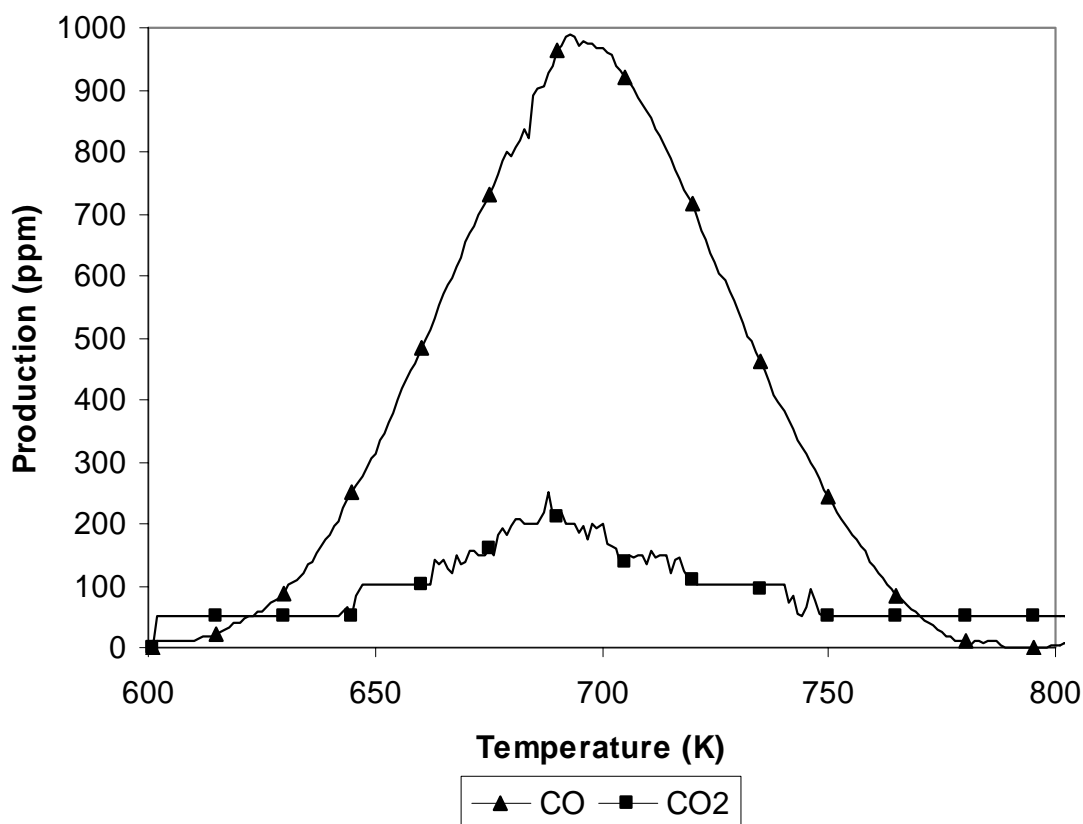


Figure 4.4: CO and CO₂ production of 59.4% n-decane/40.6% iso-octane mixture.

Thus, the mixture entered the NTC region within 4 K of the 690 K measured for the Fischer-Tropsch jet fuel. Since the mixture produced 100 ppm more CO than the Fischer-Tropsch jet fuel, a slight adjustment of the mixture composition will be required to properly match the low temperature reactivity.

4.2.3 Oxidation of the Tuned n-Decane/iso-Octane Mixture

Figure 4.5 compares CO and CO₂ production of the Fischer-Tropsch jet fuel and the mixture of 53.1% n-decane / 46.9% iso-octane by volume. The mixture

proportions were first estimated to approximate the cetane number, and then further modified to approximate the reactivity of Fischer-Tropsch jet fuel. Maximum CO production was approximately 875 ppm for both JP-8 and the two component mixture, and identical within the experimental error of 50 ppm. The maximum CO₂ production for both fuels was also the same at approximately 200 ppm.

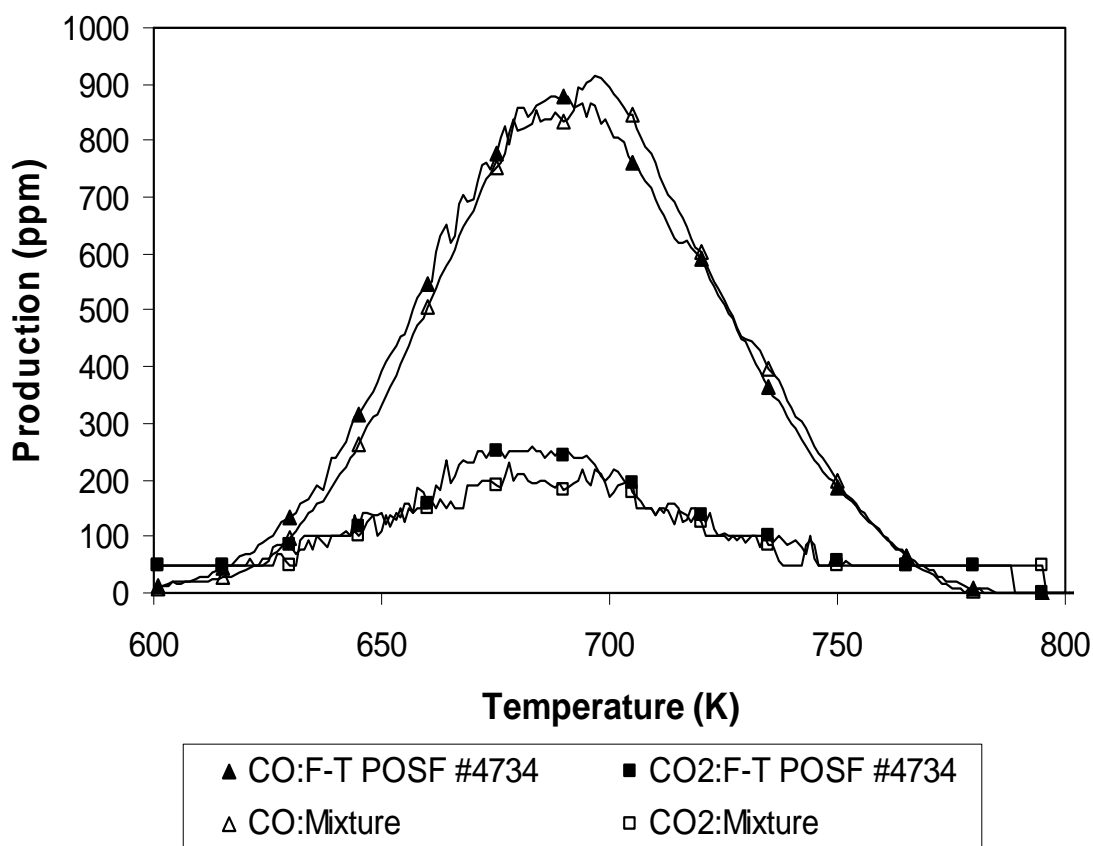


Figure 4.5: Reactivity map of Fischer-Tropsch jet fuel and 53.1% n-decane / 46.9% iso-octane mixture.

4.3 Closure

Fischer-Tropsch jet fuel was oxidized in the PFR facility at temperatures of 600-800 K and 8 atm pressure. Before entering the NTC region of reduced activity,

Fischer-Tropsch jet fuel was significantly more reactive than petroleum-derived JP-8. The higher amount of reactivity is caused by the difference in chemical composition between the two types of fuel. Fischer-Tropsch jet fuel, unlike petroleum-derived JP-8, contains 100% alkanes.

There is an ongoing effort to develop surrogate fuels to aid in developing models for the real fuels. Due to the differences in chemical composition of the fuels, such as the lack of aromatics and cycloalkanes in Fischer-Tropsch jet fuel, surrogates for petroleum-derived JP-8 may not be appropriate for Fischer-Tropsch jet fuel when reactivity and emissions are the targets for the surrogate. Such targets may be crucial for the optimization of advanced engine designs such as HCCI engines, where ignition relies on the chemistry of the fuel.

An initial mixture of 59.4% n-decane/40.6% iso-octane was investigated as a possible surrogate for Fischer-Tropsch jet fuel, with the targets being chemical formula and reactivity in the low and intermediate temperature regimes. This mixture produced more CO than Fischer-Tropsch jet fuel; therefore, a slight adjustment of the mixture's composition, 53.1% n-decane/46.9% iso-octane, was required to properly match the low temperature reactivity based on the production of CO.

CHAPTER 5: THE OXIDATION OF ALTERNATIVE JET FUELS AND A FISCHER-TROPSCH JET FUEL SURROGATE IN COMPLEMENTARY COMBUSTION FACILITIES

5.1 Introduction

The purpose of this phase of the study was to explore the reactivity differences and to investigate the preignition and autoignition behavior of alternative jet fuels compared to petroleum jet fuel in a pressurized flow reactor (PFR) and single cylinder research engine. Results were compared to a typical sample of petroleum derived jet fuel, POSF 3773. A possible surrogate for Fischer-Tropsch jet fuel, which contains a mixture of 53.1% n-decane/46.9% iso-octane, was also tested in the facilities to compare preignition and autoignition behavior. The alternative jet fuels selected were the Fischer-Tropsch jet fuel used in Chapter 4 and a coal derived jet fuel identified as POSF 4765. Both alternative fuels are manufactured from sources other than petroleum. Table 5.1 shows properties of the coal derived and Fischer-Tropsch jet fuels. The experimental conditions are presented in Section 5.2. Section 5.3 discusses the results of the PFR and research engine experiments. Section 5.4 closes the Chapter.

Table 5.1: Properties of coal derived and Fischer-Tropsch jet fuel samples.

Property	Coal POSF 4765	Fischer-Tropsch POSF 4734
Aromatics	1.8% vol	0.0% vol
Alkenes	-	-
Cetane Index	-	-
API Gravity	33.1	55.3
Heat of Combustion	43.0 MJ/kg	44.1 MJ/kg
Hydrogen Content	13.2% mass	15.3% mass
Fuel System Icing Inhibitor	-	-
Total Sulfur	0.00% mass	0.00% mass
IBP Distillation	181 °C	153 °C
10% Distillation	192 °C	169 °C
20% Distillation	194 °C	176 °C
50% Distillation	204 °C	201 °C
90% Distillation	243 °C	249 °C
EP Distillation	270 °C	271 °C

5.2 Experimental Conditions

For the PFR experiments, the controlled cool down methodology, as described in Section 3.4, was followed. Engine experiments were conducted with the Cooperative Fuel Research (CFR) engine coupled to the dynamometer, as described in Section 3.3. For this phase of the study, the compression ratio was held constant at 16:1. The bore is 8.25 cm, the stroke is 11.43 cm, and the displacement is 611 cm³. For each experiment, the inlet manifold pressure was 0.1 MPa, the engine speed was 750 RPM, and the equivalence ratio was 0.478. Experiments were run at an inlet temperature of 480 K, above the boiling points of the fuels and thus eliminating concerns about fuel condensation.

5.3 Results and Discussion

All three of the fuels, plus the surrogate mixture, were oxidized in both facilities. The order of reactivity in descending order was Fischer-Tropsch, petroleum, and coal derived jet fuel. The reactivity differences are attributed to composition differences. The results from the PFR are presented first, followed by the results from the single cylinder research engine.

5.3.1 Jet Fuels in the PFR

Three samples of jet fuel were oxidized in the PFR. As the actual empirical formula was unknown and was not determined for this study, an average formula of $C_{11}H_{21}$ was used (Edwards et al., 2001). This led to a fuel flow rate into the PFR of 1.050 mL/min at the given equivalence ratio and nitrogen dilution of fuel. The same fuel flow rate was used for the other jet fuel samples. This fuel flow rate at the given equivalence ratio and nitrogen dilution of fuel corresponds with a fuel concentration of 775 ppm, and thus an average carbon atom concentration of 8525 carbon atoms per million molecules of reacting mixture (fuel, N_2 , and O_2) was available. As shown in Fig. 5.1, the petroleum JP-8 showed substantial reactivity (>150 ppm CO) at 644 K. Reactivity continued to increase as temperature increased, until peak reactivity, indicated by 650 ppm CO, occurred at 689 K. The characteristic NTC region then occurred, as reactivity decreased with increasing temperature. Significant reactivity ceased at 742 K.

The reactivity of the natural gas derived jet fuel, POSF 4734, was different from the petroleum derived, POSF 3773, sample. The Fischer-Tropsch jet fuel sample showed significant reactivity at temperatures as low as 632 K, 12 K lower

than the petroleum sample. The Fischer-Tropsch sample produced a maximum of 880 ppm CO at 687 K. The NTC peaks occurred at approximately the same temperature, but the Fischer-Tropsch jet fuel produced 230 ppm CO more than the petroleum sample. Significant reactivity continued until 754 K, 12 K higher than the petroleum sample. Thus, the Fischer-Tropsch jet fuel showed reactivity across a wider range of temperatures, and this range was symmetrical when comparing the Fischer-Tropsch jet fuel to the petroleum JP-8, as shown in Fig. 5.1.

The composition can explain the difference in reactivity of the Fischer-Tropsch jet fuel. The Fischer-Tropsch process produces a fuel of n- and iso-paraffins. Such fuels have cetane numbers of approximately 70 or higher, while petroleum JP-8 has cetane numbers of approximately 45 (Muzzell et al., 2006). These large differences in composition and cetane number have a significant effect on preignition reactivity.

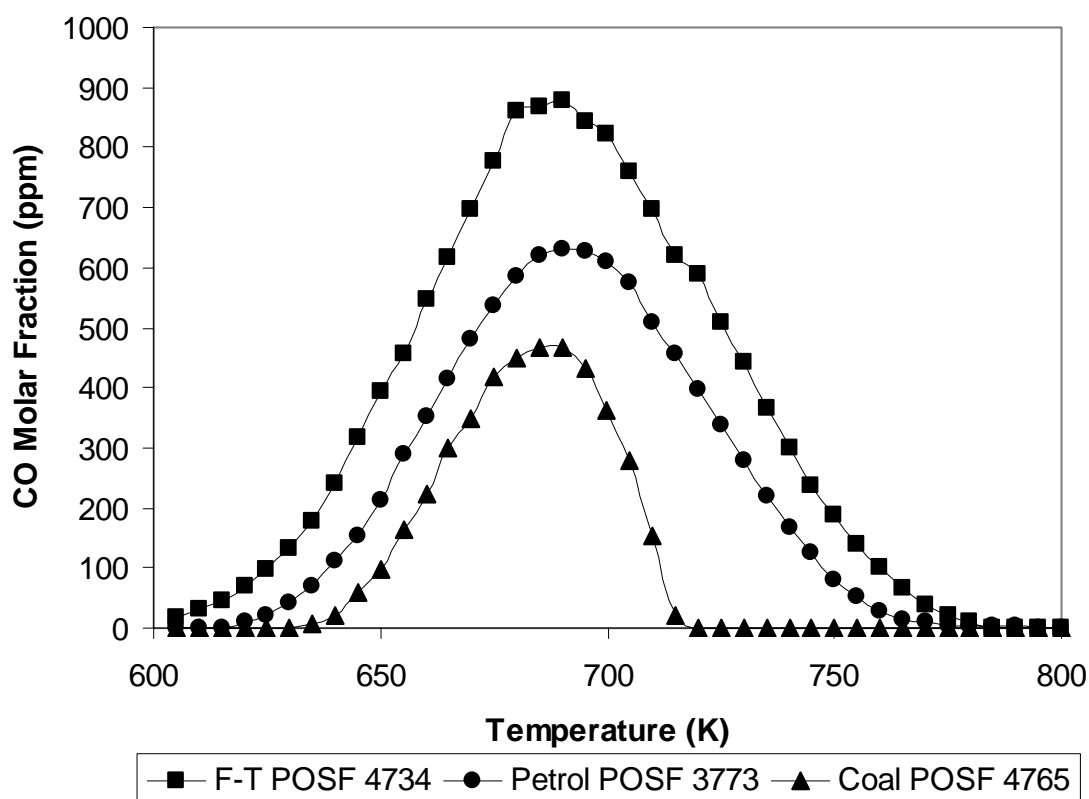


Figure 5.1: Reactivity map of natural gas, petroleum, and coal derived jet fuel in the pressurized flow reactor.

The reactivity of the coal derived jet fuel was different from both the petroleum and natural gas derived jet fuels. Figure 5.1 shows that the coal derived jet fuel produced significantly less CO than the other samples and had a much narrower low temperature reaction zone. Significant reactivity did not begin until 655 K, 11 K higher than for the petroleum sample. The coal jet fuel produced a maximum of 480 ppm CO at 688 K. While the temperature of peak reactivity was approximately the same as the other samples, the value for the coal jet fuel was lower, producing 170 ppm CO less than the petroleum sample. Notably, the reaction region of the coal jet fuel was not only narrower, as shown in Fig. 5.1, but also asymmetric with the

reactivity decreasing more sharply with increasing temperature. Significant reactivity was observed at temperatures only up to 710 K, 32 K less than the petroleum sample.

The reactivity differences of the coal-derived jet fuel again can be attributed to the composition. Previous PFR experiments, by Agosta (2002), of methylcyclohexane and decalin have shown that cycloparaffins have narrow ranges of reactivity at temperatures of 600-800 K. Two decalin experiments were conducted in the PFR. The first decalin experiment had an equivalence ratio of 0.3 and CO was produced between the narrow region of 656-724 K. For the second decalin experiment the equivalence ratio was increased to 0.4. For this experiment CO was produced between the temperature range of 651-726 K. However, even though decalin had a narrow CO reactivity map, CO was produced at a high rate, at the start of reaction. Since cycloparaffins represent approximately 20 % of the chemical composition of this JP-8, the narrow CO reactivity map of JP-8 is expected (Agosta, 2002).

5.3.2 Petroleum JP-8, Coal Derived Jet Fuel, and Fischer-Tropsch Jet Fuel Surrogate in the Single Cylinder Research Engine

The jet fuels were also tested in the single cylinder research engine. Figure 5.2 shows data from the autoignition regime of the engine cycle, the data portion relevant for this study. The horizontal-axis refers to the crank angle degrees with 360° being top dead center for compression. The three fuel samples are the same as those run in the PFR. A motored run with no fuel is also shown to display the in-cylinder pressure rise due to compression of air. The results are an average of three

runs. The onset of combustion is shown in Fig. 5.2 as the knee in the pressure trace. For the Fischer-Tropsch jet fuel, the onset of combustion occurred at 339 CAD. The onset of combustion for the petroleum JP-8 was 342 CAD. In both fuels, first stage ignition reactivity is observed, as the fuels exhibit extended periods of increasing pressure as CAD increases, from the onset of the combustion until peak pressure at approximately 353 CAD. The coal jet fuel, however, showed different autoignition behavior. The onset of combustion occurred at 349 CAD, 7 CAD later than the petroleum JP-8. Additionally, the coal jet fuel did not reach peak pressure until 358 CAD, approximately 5 CAD later than the other fuels.

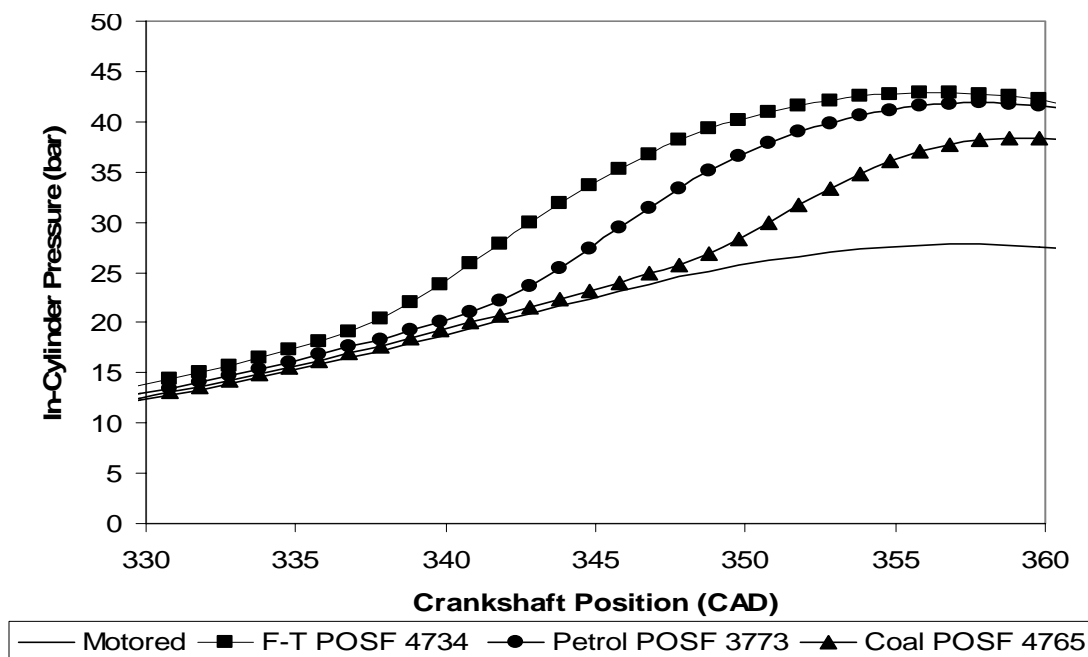


Figure 5.2: Autoignition of natural gas, petroleum, and coal derived jet fuel in the single cylinder research engine.

Thus, the duration from onset of combustion to peak pressure, which can be considered an indication of first stage ignition reactivity, was longer for the Fischer-

Tropsch and petroleum jet fuels than for the coal jet fuel. The former two samples had CAD durations of 14 CAD and 11 CAD, respectively, while the latter exhibited a duration of 9 CAD. This preignition reactivity behavior is consistent with the PFR results. In the PFR, the Fischer-Tropsch jet fuel produced the maximum CO at its maximum level of reactivity in the low temperature regime, followed by the petroleum JP-8 and lastly the coal jet fuel. These experiments show the importance of low temperature chemistry in the autoignition process of engines.

Previously in Chapter 4, a surrogate was developed for natural gas-derived Fischer-Tropsch jet fuel and tested it in the PFR. The surrogate is composed of 53.1% n-decane / 46.9% iso-octane (2,2,4-trimethylpentane) by liquid volume and matched the average composition and low temperature reactivity of Fischer-Tropsch jet fuel, POSF-4734, in the PFR. To further explore this surrogate, the surrogate was tested in the single cylinder research engine and the results are presented in Fig. 5.3. The surrogate matched the real fuel fairly well. The onset of combustion for the former was 340 CAD, 1 CAD later than the JP-8. By 344 CAD and later, both the surrogate and Fischer-Tropsch jet fuel showed matching behavior, and they both reached peak pressure at 353 CAD.

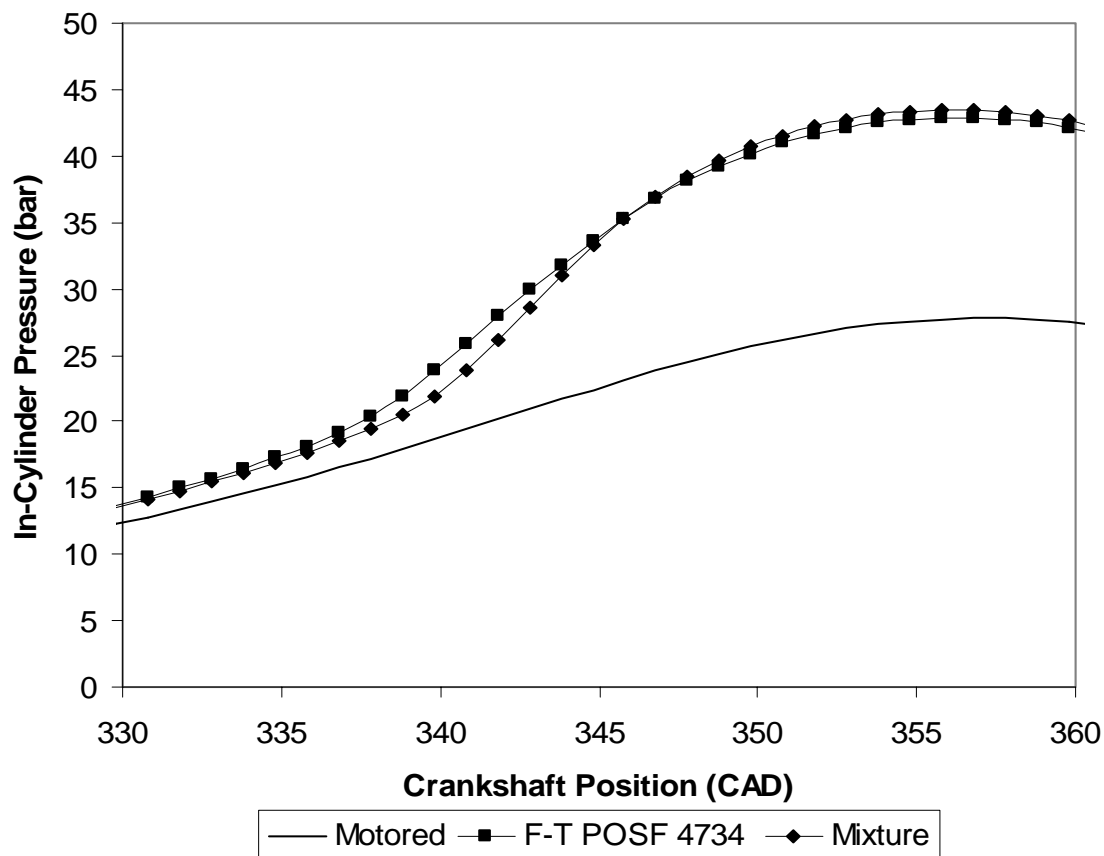


Figure 5.3: Autoignition of surrogate for Fischer-Tropsch jet fuel.

5.4 Closure

This chapter presented experimental results from the oxidation of petroleum derived JP-8, natural gas derived Fischer-Tropsch jet fuel, coal derived jet fuel, and a surrogate mixture for Fischer-Tropsch jet fuel in complementary facilities. The reactivity behavior of these possible future fuels differs from that of traditional petroleum derived JP-8. The feedstocks and processing methods of the alternative fuels lead to differences in composition. At preignition and autoignition conditions, such composition differences can have large effects. Future engine designs will have to consider these differences. The order of reactivity in descending order was

Fischer-Tropsch, petroleum, and coal. In the engine, the surrogate matched autoignition behavior of Fischer-Tropsch jet fuel.

CHAPTER 6: INTERMEDIATE SPECIES ANALYSIS OF N-DECANE AND DECALIN

6.1 Introduction

The purpose of this phase of the study was to examine the intermediate species produced during the low and intermediate temperature oxidation of neat n-decane and decalin in the pressurized flow reactor facility, to identify the branching pathways controlling auto-ignition. Section 6.2 discusses the experimental conditions and Section 6.3 presents the results of the analysis. The closure of this chapter is presented in Section 6.4.

As mentioned in Chapter 4, n-decane was selected as a component representing the linear alkanes of Fischer-Tropsch jet fuel because it is one of the more common high molecular weight alkanes, and it approximates the average chemical formula for JP-8, $C_{11}H_{21}$. n-Decane and iso-octane are potential surrogate components for the natural gas-derived Fischer-Tropsch jet fuel, which is almost entirely n- and iso-paraffins. n-Decane is a major component of the fuel; iso-octane is not because the branched paraffins found in the JP-8 are heavier and less branched (Bruno et al., 2006). Nevertheless, iso-octane is more readily available than the iso-paraffins found in the Fischer-Tropsch jet fuel and since iso-octane is the primary reference fuel for gasoline, there have been extensive studies and experimental validation of models pertaining to its oxidation. For instance, a detailed chemical kinetic mechanism has been developed for the oxidation of iso-octane that ranges from 550-1700 K and 0.1-4.5 MPa (Curran et al., 2002). However, while there are n-decane models available (Lindstedt et al., 2000; Bikas et al., 2001; Zhao et al., 2005),

more experimental data are needed for understanding the low and intermediate temperature chemical kinetics. A chemical analysis found that the most prevalent component of the coal-derived JP-900 POSF-4765 is trans-decahydronaphthalene (trans-decalin), followed by cis-decalin (Smith et al., 2007). To study the dicyclopaffin components in the coal-derived jet fuel, decalin (a mixture of both cis-decalin and trans-decalin) was also oxidized in the PFR.

Ultimately, the branching pathways will be useful for the improvement and advancement of the chemical mechanisms for these hydrocarbons in the low and intermediate temperature regime at elevated pressure.

6.2 Experimental Conditions

For the PFR experiments, the controlled cool down methodology as described in Section 3.4, was followed. During the CCD experiments, samples were stored in the heated sample storage cart for offline analysis, as described in Section 3.7, with the gas chromatograph with a flame ionization detector and coupled to the mass spectrometer.

6.3 Results and Discussion

n-Decane and decalin were oxidized in the PFR to analyze intermediate species produced and identify branching pathways. The n-decane results are presented first, followed by the decalin results.

6.3.1 Detailed Speciation of Intermediates Produced during n-Decane Oxidation in the PFR

Neat n-decane was more reactive in the low and intermediate temperature regime, based on CO production, than Fischer-Tropsch jet fuel and the 53.1% n-decane/46.9% iso-octane mixture, as shown in Figure 6.1. n-Decane produce a maximum CO concentration of approximately 1800 ppm, while Fischer-Tropsch jet fuel and the 53.1% n-decane/46.9% iso-octane mixture was half as reactive.

The n-decane reactivity map in Figure 6.2 shows the CO and CO₂ production and the sample extraction points for the off-line analysis. CO was produced at significant levels at temperatures as low as 615 K. The maximum CO production was approximately 1800 ppm at 700 K. Additionally, a negative temperature coefficient (NTC) region is observed from 700-800 K. Reactivity ceased at approximately 795 K.

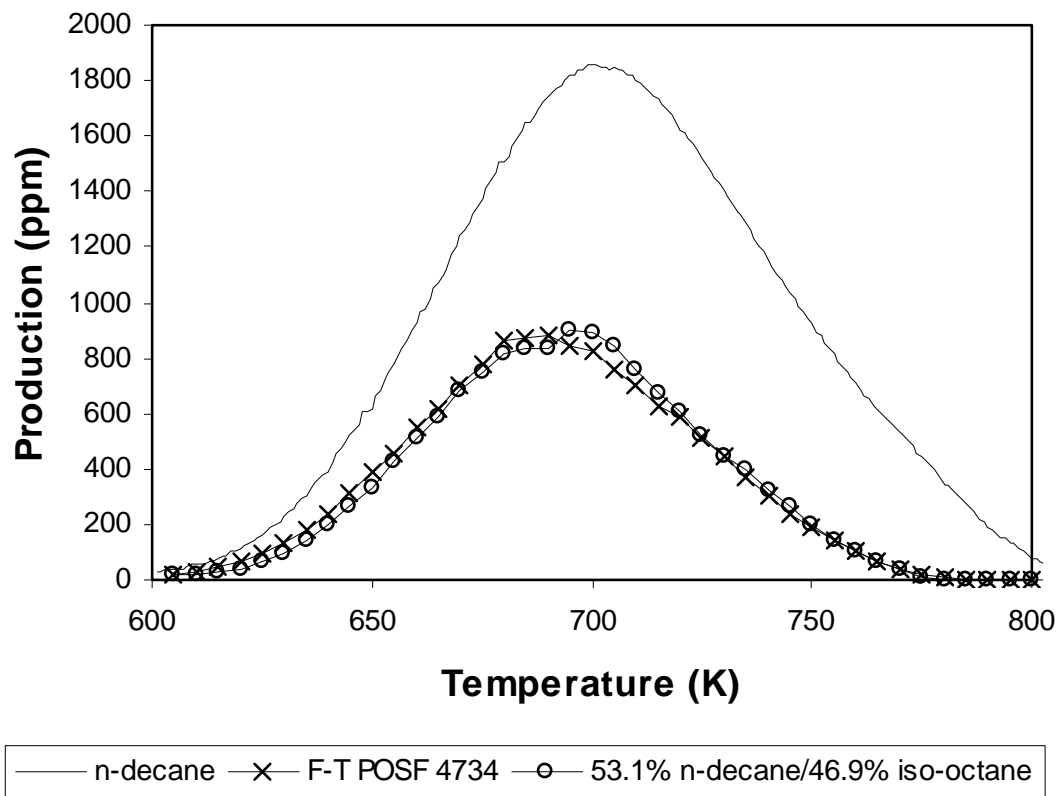


Figure 6.1: Reactivity of n-decane, Fischer-Tropsch jet fuel, and n-decane/iso-octane blend based on CO production.

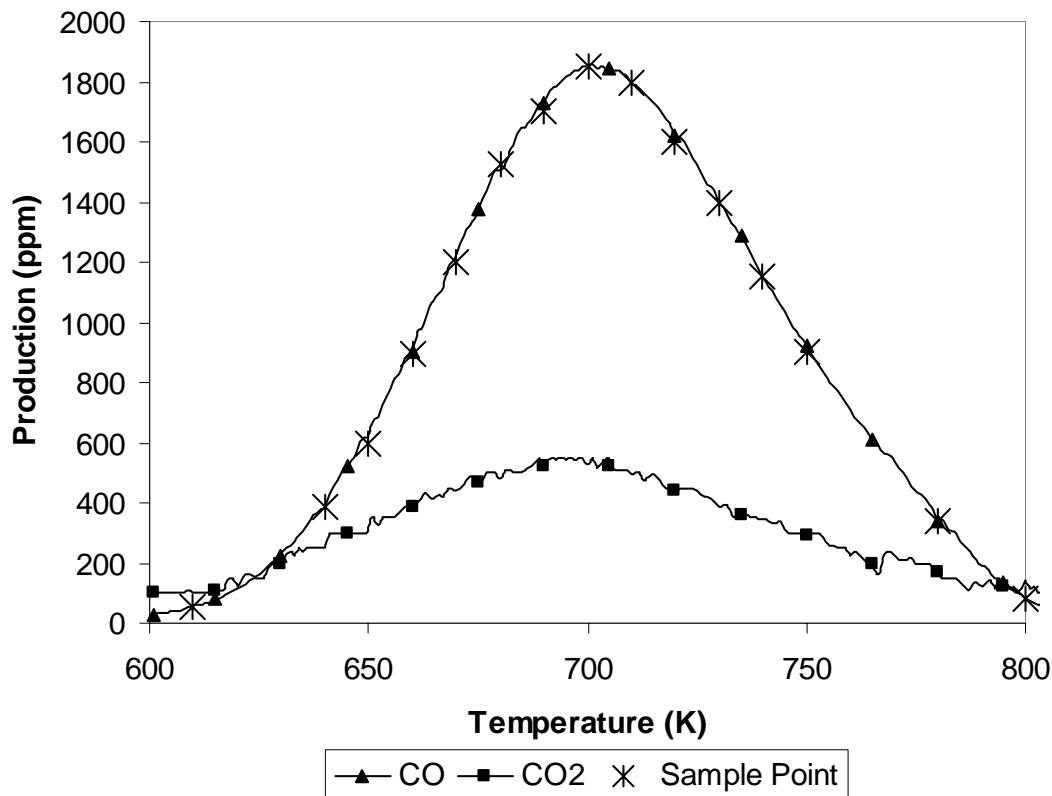


Figure 6.2: CO and CO₂ production during n-decane oxidation.

The collected samples were analyzed using the GC/MS/FID techniques described in Chapter 3. Figure 6.3 is the flame ionization detector chromatogram for the sample extracted at 700 K. Each sharp symmetrical peak is a component that was in the sample. Several major intermediate species could be positively identified. Table 6.1 lists the species and their associated elution times. The aldehydes were propanal, butanal, pentanal, hexanal, and heptanal. Formaldehyde is to be expected as a major intermediate. However, it was not identified using the current GC/MS parameters. The alkenes included ethene, propene, 1-butene, 1-pentene, and 1-hexene. The major alkane was n-decane. Two decene isomers, 5-decene and 4-decene, were also identified. Additionally, 2-butanone and 2-pentanone and several cyclic ethers were

also identified. Figure 6.4 shows the chemical structure of species that were produced during the oxidation of n-decane.

Table 6.1: Species identified by GC/MS/FID in n-decane oxidation samples.

Species	Elution time (min)
Ethene	9.89
Propene	11.02
Butene	13.67
Propanal	16.76
1-Pentene	17.18
Butanal	19.88
2-Butanone	20.13
1-Hexene	20.33
2-Pentanone	22.51
Pentanal	22.70
Hexanal	25.25
Heptanal	27.78
5-Decene	30.55
4-Decene	30.62
n-Decane	30.77

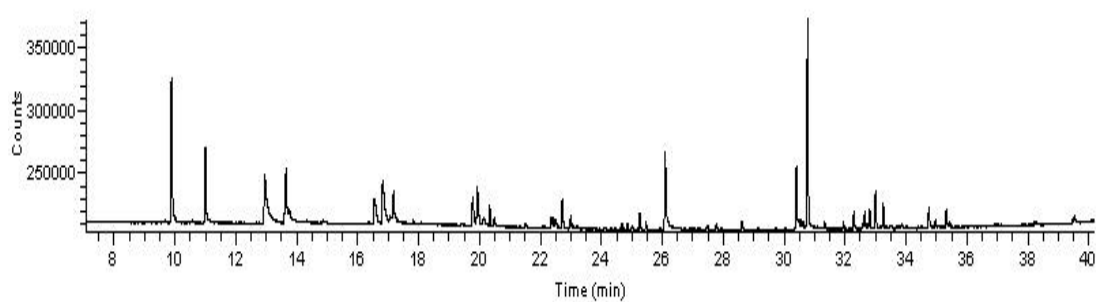


Figure 6.3: Chromatogram of the 700 K n-decane oxidation sample.

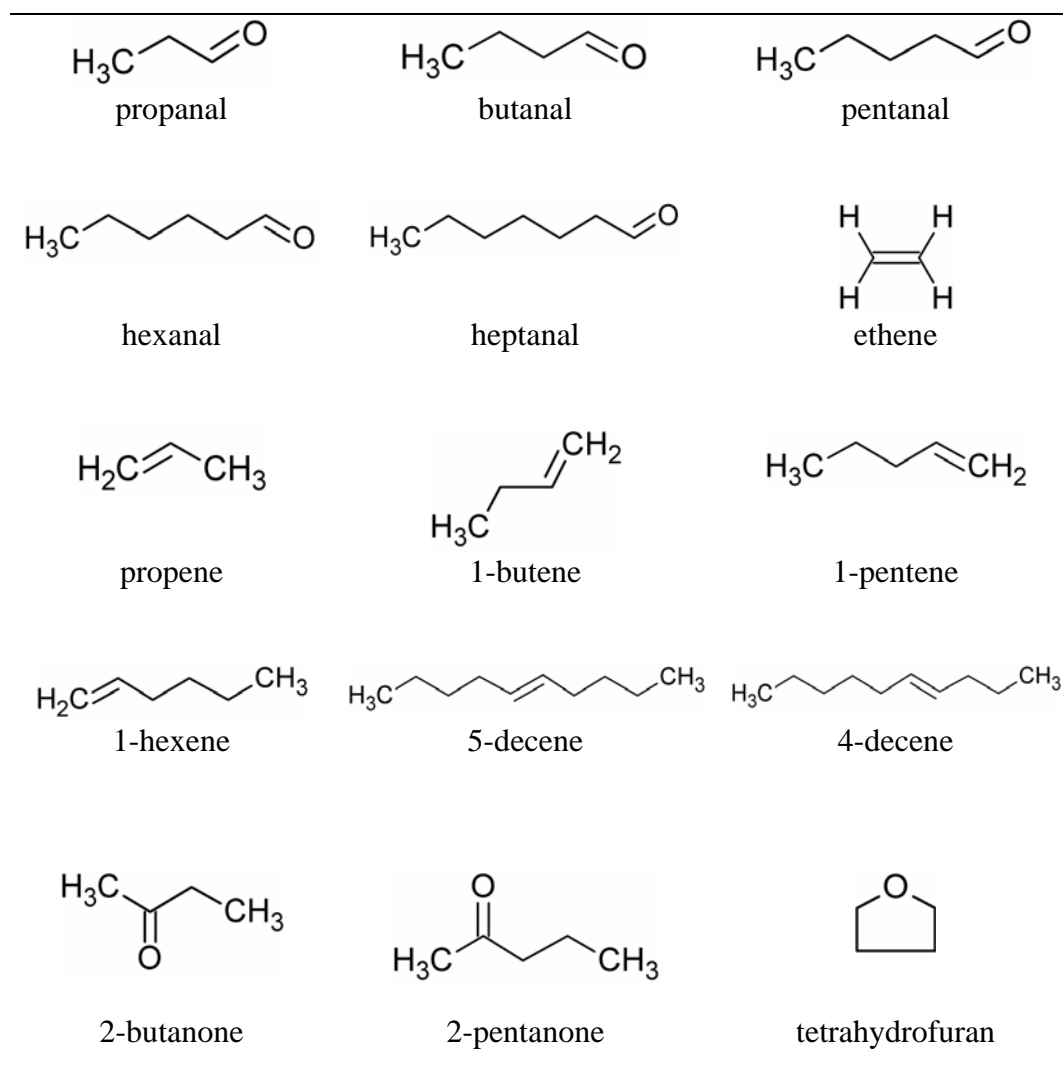


Figure 6.4: Chemical structure of the identified species during n-decane oxidation.

6.3.2 Detailed Speciation of Intermediates Produced during Decalin Oxidation in the PFR

Figure 6.5 shows production of CO and CO₂ from the oxidation of decalin in the low and intermediate temperature regime. Significant reactivity was observed at temperatures from 626 K to 731 K. Thus, decalin has a limited range of reactivity at low temperatures compared to paraffins such as n-decane. Reactivity peaked with

2010 ppm CO at 692 K. CO₂ was also measured and showed similar trends as CO production but with approximately 25% of the quantity.

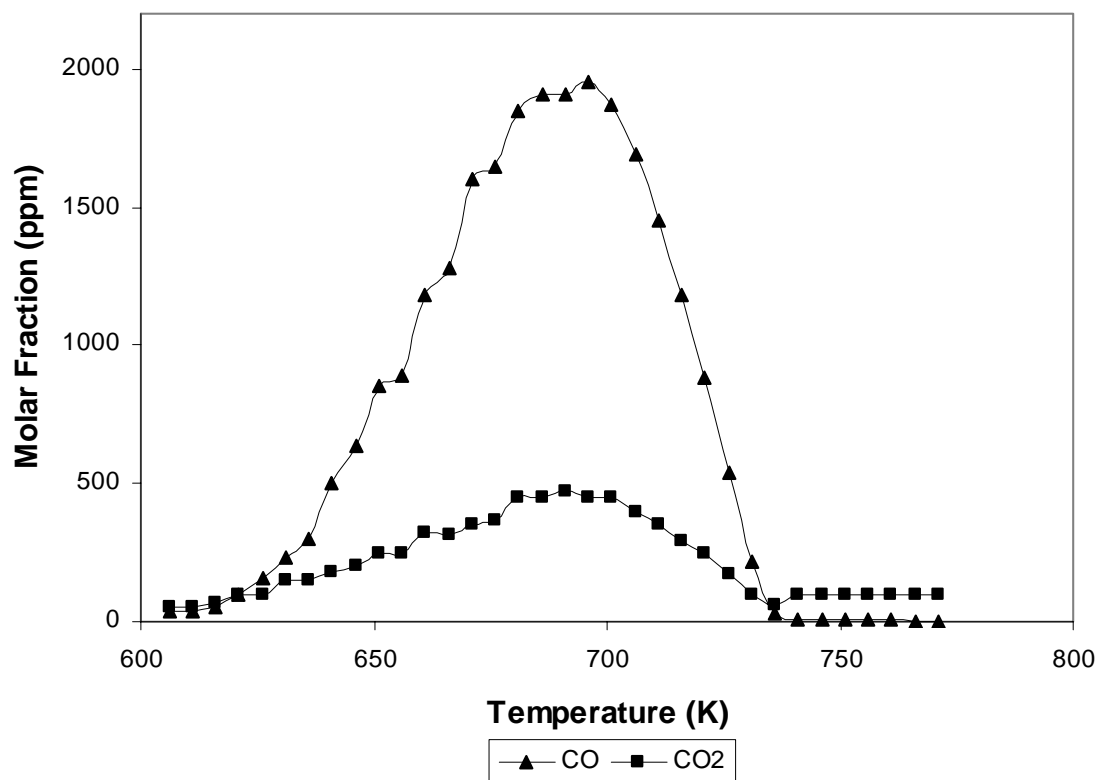


Figure 6.5: Reactivity map of decalin oxidation in the PFR.

During the decalin experiment, samples were also stored in the heated sample storage cart and analyzed with the GC/MS/FID system. Table 6.2 shows the species identified along with their elution times and the temperatures at which they were identified. Sixteen species were identified (not including CO and CO₂ quantified with the NDIR analyzer). Several different classes were represented, including light alkenes (propene and 2-butene), light aldehydes (acetaldehyde, 2-propenal, and butanal), monocycloalkenes (cyclohexene and 1-methylcyclohexene), ketone-substituted monocycloalkenes (2-cyclohexenone), aromatics (benzene and toluene),

monocycloalkanes (methylcyclohexane and ethylcyclohexane), ketone-substituted monocycloalkanes (cyclopentanone and cyclohexanone), and dicycloparaffins (cis- and trans-decalin). The decalin sample for the experiment was a mixture of cis- and trans-decalin, as purchasing the individual isomers is much more expensive.

Nevertheless, the isomers were successfully separated and identified. At 773 K, a sample was collected with only the prevaporized nitrogen-fuel mixture and the nitrogen component of the synthetic air flowing in the PFR. Small traces of methylcyclohexane and ethylcyclohexane were identified at this point and may be attributed to the fuel sample. Figure 6.6 shows the chemical structure of each species that were identified.

Table 6.2: Species identified by GC/MS/FID in decalin oxidation samples.

Species	Elution time (min)	648 K	690 K	726 K	773 K (no O ₂)
Propene	10.96	X	X	X	
Acetaldehyde	13.02	X	X		
2-Butene	13.63			X	
2-Propenal	16.65	X	X	X	
Butanal	19.98	X	X	X	
Benzene	22.09	X	X	X	
Cyclohexene	22.78	X	X	X	
Methylcyclohexane	24.00	X	X	X	X
Toluene	24.82	X	X	X	
Cyclopentanone	24.99	X	X		
1-Methylcyclohexene	25.11		X	X	
Ethylcyclohexane	26.83	X	X	X	X
Cyclohexanone	27.71	X			
2-Cyclohexenone	28.59	X			
trans-Deca- hydronaphthalene	32.93	X	X	X	X
cis-Deca- hydronaphthalene	34.16	X	X	X	X

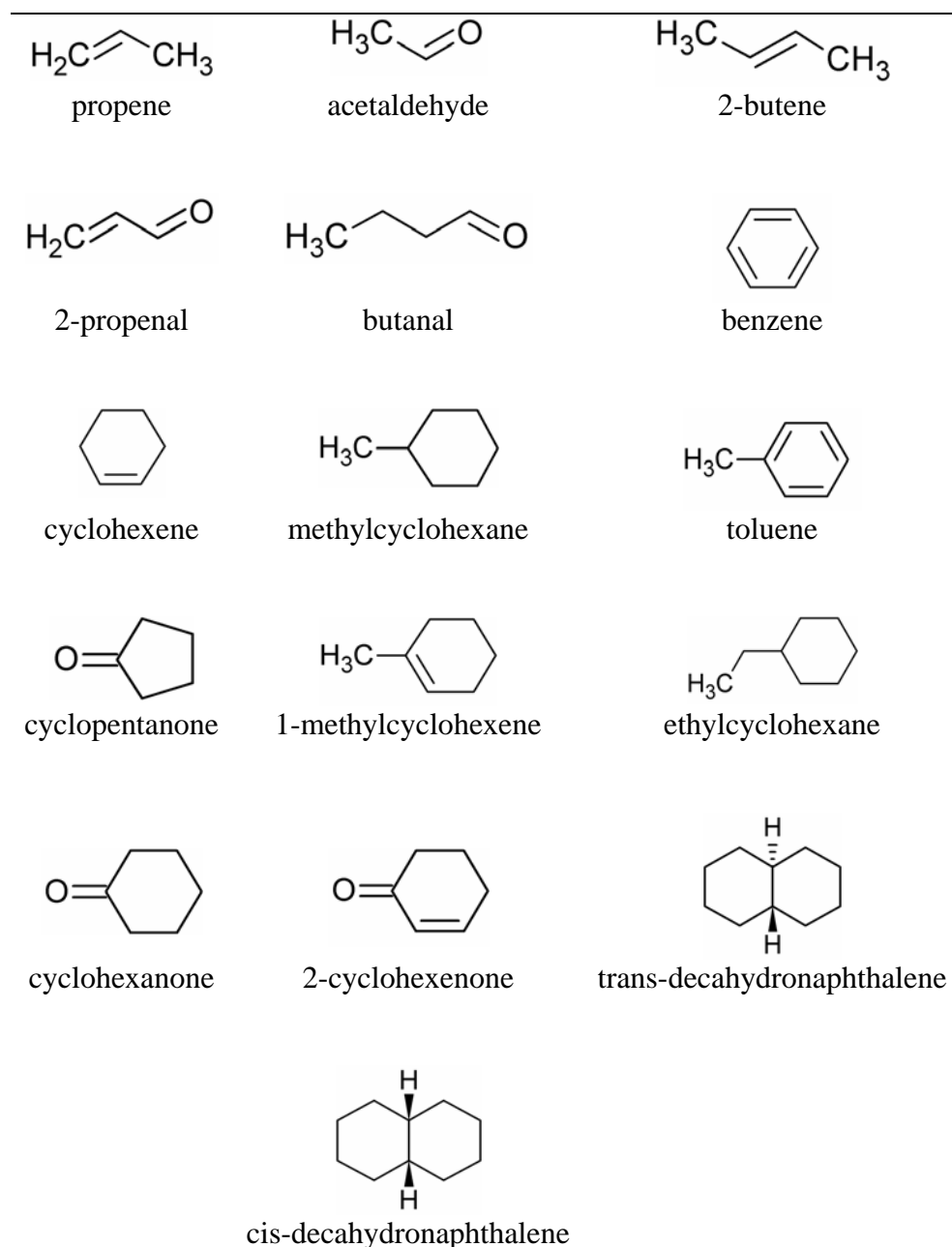


Figure 6.6: Chemical structure of the identified species during decalin oxidation.

Figure 6.7 shows a sample chromatogram, from 690 K. The two large spikes show the two decalin isomers. All other species identified were in much smaller quantities, and thus they appear absent from the chromatogram in Fig. 6.7 because the graphs display relative abundance. Nevertheless, the information presented in Table 6.2

shows some interesting aspects of the low temperature oxidation of decalin. At the lowest temperature at which a sample was taken, 648 K, two ketones were identified that were not present at the higher temperatures. These species were cyclohexanone and 2-cyclohexenone. Cyclohexanone has a molecular formula of $C_6H_{10}O$; it is cyclohexane with one substituted ketone group. Ketones can be produced from several reactions in the low and intermediate temperature regime. One possible pathway is hydroxyl radical addition to an alkene that produces a hydroxyalkyl radical, followed by isomerization and hydrogen abstraction to produce the ketone (Touchard et al., 2005). 2-Cyclohexenone has a molecular formula of C_6H_8O ; it is cyclohexene with one substituted ketone group. Other cycloparaffins identified included cyclohexene, cyclopentanone, and 1-methylcyclohexene. Of those, one is an additional ketone-substituted cycloparaffin, cyclopentanone. This molecule was only identified at the two lower temperatures, 648 K and 690 K. Thus, the presence of the ketone-substituted cycloparaffins at the lower temperatures, and none at the higher temperature of 726 K, shows that the pathway to ketone substitution is present only at lower temperatures before the NTC region dominates.

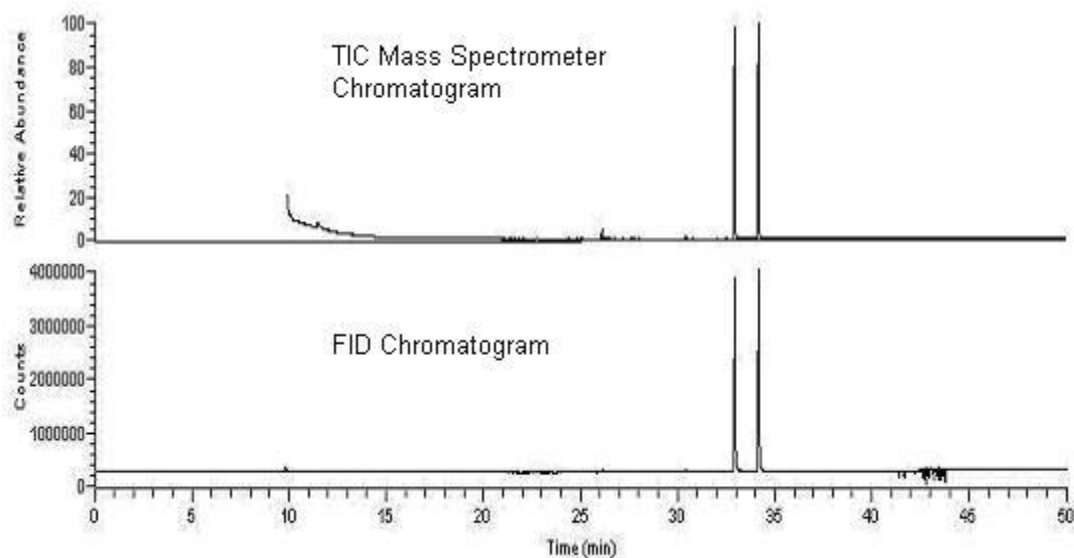


Figure 6.7: MS (top) and FID (bottom) chromatograms produced from decalin oxidation at 690 K.

A number of C_6 cycloparaffin components were identified. Clearly there was cleavage of the two cyclohexane components of the parent fuel decalin. Accompanying this, there were several C_4 non-cyclic species – butanal and 2-butene. No C_4 cyclic species, such as cyclobutanone, were identified, suggesting that when decalin decomposes to produce a C_6 cyclic species, the other portion of the parent fuel will not maintain the cyclic structure. Furthermore, the presence of cyclopentanone, a C_5 cycloparaffin, suggests the possibility that decalin decomposition can lead to two C_5 cyclic species. The other possible method of cyclopentanone production, from C_6 cycloparaffins, is less likely (Lenhert, 2004b). The previous study of methylcyclohexane oxidation in the PFR facility did not identify any C_5 cyclic species.

Decalin has 10 carbon atoms, including 8 secondary and 2 tertiary. Tertiary C-H bond types, those on carbon atoms that are bonded to three other carbons atoms,

are weaker than secondary C-H bonds, which are on carbon atoms that are bonded to two other carbon atoms. From the species identified in this study, it appears that the tertiary bonds are more likely to break, even though they are outnumbered 4:1 by the secondary bond types.

Additionally, two aromatic species - benzene and toluene – were identified. As no C₆ alkenes were identified, it is likely that these species were produced from the C₆ cyclic species through a direct dehydrogenation process.

6.4 Closure

This chapter presented detailed experimental results, including intermediate species identification, from the oxidation of neat n-decane and decalin throughout the low and intermediate temperature regime in the PFR. During the PFR experiments, samples were extracted at various temperatures for GC/MS/FID analysis. Identifying the branching pathways will be useful for the improvement and advancement of the chemical mechanisms for these hydrocarbons.

n-Decane was selected as a component of Fischer-Tropsch jet fuel surrogate because it is one of the more common high molecular weight alkanes, and it approximates the average chemical formula for JP-8. Neat n-decane was more reactive than Fischer-Tropsch jet fuel and the 53.1% n-decane/46.9% iso-octane mixture. The major intermediate species that were identified included components of the functional groups aldehydes and alkenes. Additionally, 2-butanone and 2-pentanone and several cyclic ethers were also positively identified.

The most prevalent component of the coal-derived JP-900 POSF-4765 is trans-decalin, followed by cis-decalin. To study the dicycloparaffin components in

the coal-derived jet fuel, decalin was also oxidized in the PFR. Decalin reactivity exhibited a limited range of reactivity at low temperatures compared to paraffins such as n-decane. Alkenes and aldehydes were also present as products of combustion. Additionally, many cyclic species were also identified.

With moderate and high molecular weight hydrocarbon fuels, identifying and quantifying the large carbon number intermediates is a difficult problem that generally leads to not all of the carbon being counted. Work is underway to examine the possibility of condensation in the sampling system and reaction in the storage cart.

CHAPTER 7: CONCLUSIONS AND RECOMMENDATIONS FOR FUTURE WORK

7.1 Introduction

This study investigated the oxidation of alternatives for petroleum derived JP-8 in a pressurized flow reactor and a single cylinder research engine. The alternative fuels investigated were a natural gas derived Fischer-Tropsch jet fuel and a coal derived jet fuel produced by a hydrotreating/hydrogenation process. In order to identify the branching pathways controlling autoignition, intermediate species produced during the low and intermediate temperature oxidation of neat n-decane and decalin were also investigated.

The pressurized flow reactor was used to evaluate the oxidation characteristics of the hydrocarbons under lean, dilute conditions at temperatures of 600-800 K and 8 atm pressure. Experiments were also conducted in a single cylinder research engine to examine autoignition of the fuels. Dissimilarities in autoignition and measured intermediates were observed between the petroleum and alternative jet fuels, which are consistent with the known differences in chemical composition; therefore, different surrogates need to be developed for each alternative fuel.

This research will aid in the overall development and use of chemical kinetic models that are necessary to simulate combustion characteristics of gas turbines and CI engines operating with these fuels. The simulations will aid in efforts to improve fuel efficiency, emissions, and power output. The results of this research are summarized in this chapter. Recommendations for future work are also discussed.

7.2 Conclusions

The initial component of this study was to investigate Fischer-Tropsch jet fuel, compare its reactivity to petroleum JP-8, and propose surrogate components. The alternative fuel was oxidized in the PFR facility at temperatures of 600-800 K and 8 atm pressure. Fischer-Tropsch jet fuel was significantly more reactive than petroleum-derived JP-8 before entering the NTC region. Due to the lack of aromatics and cycloalkanes in Fischer-Tropsch jet fuel, surrogates for petroleum derived JP-8 research are not applicable for Fischer-Tropsch jet fuel when chemical composition and low temperature reactivity are targets for the surrogate. Such a target may be crucial for the optimization of advanced engine designs such as Premixed Compression Ignition (PCI) engines, where ignition relies on the chemistry of the fuel. In anticipation of the slightly higher cetane number of Fischer-Tropsch jet fuel compared to petroleum derived JP-8, a mixture of 59.4% n-decane and 40.6% iso-octane was investigated as a possible surrogate for Fischer-Tropsch jet fuel, with the targets being chemical composition and reactivity in the low and intermediate temperature regimes. Matching only cetane numbers between the mixture and Fischer-Tropsch jet fuel is not sufficient in predicting the overall low and intermediate temperature reactivity of the fuel.

The next phase of the study investigated the n-decane/iso-octane mixture in the PFR facility at temperatures of 600-800 K and 8 atm pressure. Several experiments were performed with the binary mixture to evaluate and appropriately tune the mixture to represent the reactivity, based on CO production, throughout the low and intermediate temperature regime. The binary mixture of 53.1% n-decane and

46.9% iso-octane represented the CO production of Fischer-Tropsch jet fuel under these conditions.

The study then explored the reactivity differences of alternative jet fuels compared to petroleum jet fuel, and their surrogate components in two complementary combustion facilities. Preignition experiments were conducted in the pressurized flow reactor and autoignition experiments were conducted in the single cylinder research engine. The alternative jet fuels were a coal-derived jet fuel and a natural gas-derived Fischer-Tropsch jet fuel. In both facilities, the order of reactivity in descending order was Fischer-Tropsch, petroleum, and coal. The reactivity differences are attributed to composition differences.

Possible surrogates and their components were also tested in the facilities to elucidate how compositional differences affect preignition and autoignition chemistry. Neat n-decane, the dominant species in the Fischer-Tropsch jet fuel, and neat decalin, which is the most prevalent component of the coal derived jet fuel sample, were investigated in the PFR. Samples were extracted throughout the low and intermediate temperature regime for GC/MS analysis, in order to identify the branching pathways controlling auto-ignition. The major intermediate species that were identified for n-decane oxidation included components of the functional groups aldehydes and alkenes. The major intermediate species that were identified for decalin oxidation included components of the functional groups aldehydes, alkenes, cyclo-alkenes, cyclo-ketones, and one aromatic, benzene. The pathways will ultimately be useful for the improvement and advancement of the chemical

mechanism for n-decane and decalin in the low and intermediate temperature regime at elevated pressure.

Overall, the research presented will aid in the development of chemical kinetic models for non-petroleum derived jet fuel that will be employed for simulating combustion characteristics of gas turbines and CI engines while improving such traits as fuel efficiency, emissions, and power output.

7.3 Recommendations for Future Work

Significant work has been completed and presented about the understanding of the oxidation of alternative fuels for petroleum derived JP-8. However, the combustion chemistry that is occurring in the low and intermediate temperature regime is not a straightforward and simple process. With high molecular weight hydrocarbons, the oxidation chemistry is very complex. This study presents data to aid in the understanding of the complex oxidation processes. Some selected recommendations for future work are presented as follows.

(1) This research presented a qualitative analysis of species, other than CO and CO₂, produced during the oxidation of n-decane and decalin. To gain full advantage and understanding of reaction pathways that are being produced, it is also necessary to have a quantitative analysis as well. The quantitative analysis will aid in determining the rates at which species are produced. However, the problem that arises with trying to quantify some of these high molecular weight species is that a known quantity has to be analyzed in the GC/MS/FID system to develop a calibration curve. Once the calibration curve for a species is developed, the species can be quantified. The difficulty is that some high molecular weight species can only be

purchased in liquid phase samples. The liquid phase samples then have to be vaporized to gas phase in a known concentration. To maintain the samples in the gas phase with a known concentration has always been a difficult task.

Therefore, a method needs to be developed to provide a quantitative analysis of the amount of species produced during the oxidation process. The method could include a technique to preserve the high molecular weight species in gas phase to establish a calibration curve or a procedure to correlate the response from the GC/MS/FID of a known low molecular weight hydrocarbon to that of a high molecular weight hydrocarbon.

(2) The ability to acquire oxygen measurements would be important because several of the species that are produced in the low and intermediate temperature regime contain oxygen. An oxygen production profile could be established and aid in the understanding of reaction pathways that contain oxygen. Since the Ultramat 22P CO and CO₂ detector was experiencing operational problems, as mentioned in Section 3.5, it is recommended that the Ultramat 22P be replaced with the Ultramat 23, which includes an electrochemical oxygen measurement cell.

(3) The present coal derived and Fischer-Tropsch jet fuels do not meet all of the specifications to be designated as a JP-8 by the US military. However, coal derived and Fischer-Tropsch jet fuels can be mixed with petroleum derived JP-8 to satisfy current regulations and as a practical matter. For example, up to a maximum 50 % concentration of Fischer-Tropsch jet fuel can be mixed with petroleum JP-8 to maintain the minimum specific gravity specification. Therefore, evaluation and

testing of such blends to acquire preignition and autoignition data in our facilities is appropriate.

7.4 Closure

This study presented current research regarding the preignition and autoignition behavior of alternative jet fuels for petroleum derived JP-8 in a pressurized flow reactor and a single cylinder research engine. The alternative jet fuels studied were derived from natural gas and coal. Possible surrogates for the alternative fuels and their components were also tested to explain how compositional differences affect preignition and autoignition chemistry. Similarities and differences in preignition and autoignition behavior were observed between the petroleum and alternative jet fuels. The order of reactivity, based on CO production, in descending order was Fischer-Tropsch, petroleum, and coal. The reactivity differences are attributed to differences in chemical composition of the jet fuels. Therefore, different surrogates need to be developed for alternative jet fuels as well.

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APPENDIX A: GENERAL OPERATING PROCEDURE FOR THE PRESSURIZED FLOW REACTOR DURING A CONTROLLED COOL DOWN EXPERIMENT

CAUTION!!!

Read through the entire operating procedure before operating the pressurized flow reactor. Air flow needs to be flowing through the system any time a heater is on.

Air Flow Control

- 1.) Make sure there is air pressure. Around 80 psi during entire experiment.
- 2.) Turn on power supply for the controls. 2 red switches. (See Figure A.1)
- 3.) Turn on mass flow controller. (See Figure A.2)
- 4.) Turn on SV-5 switch for the air mass flow controller.
- 5.) Turn on and adjust mass flow controller to 50.5.

Heating

Note: ALWAYS make sure there is flow before turning on any heater!!!

- 1.) Turn on all 3 breakers.
- 2.) Every 10-20 minutes increase the 10 kW controller 20 °C. Stop at desired temperature or do not exceed 427 °C. To adjust temperature, press adjustment button on controller. Note: If temperature does not increase on temperature

controller, make sure the red limit light on the overtemp controller is not illuminated. If it is, press the reset button on the controller.

3.) Every 40-50 minutes increase the 3 kW heater controller 5%. Do not increase over 70%.

4.) Turn on the inlet, test, and outlet bead heaters at approximately 100 °C.

Note: Not all of the bead heaters need to be turned on at this time. The bead heaters heat up very quickly. For example: Turn on one bead heater for each inlet, test, and outlet area. Then turn that one off after some time has passed; 30 minutes. Turn on a different bead heater for the area being heated. Only turn on bead heaters that are required to reach the sample probe thermocouple temperature. Table A.1 is a sample detailed heater warm-up schedule.

5.) Open blue valve for the Pacesetter water supply.

6.) Turn on Pacesetter water heater at approximately 100 °C.

7.) Turn on heater for the transfer sample line elbow (Variac controller).

Increase slowly to 40% to reach desired temperature of approximately 200 °C.

8.) Turn on heater for the sample line (CIC Photonics controller) when the flow reactor is at approximately 200 °C. Try to stabilize the sample line at around 120 °C or desired temperature. Note: If the temperature of the sample line will not reach 120 °C, increase the percentage of the Variac controller.

The following can be done 1-2 hours before final temperature is reached.

9.) Connect the liquid nitrogen tanks (gas use) and oxygen bottles to the manifold lines.

- 10.) Open manifold and all necessary valves.
- 11.) Verify that the regulated pressures are adjusted properly. For the nitrogen gas: 300 psi regulated/350 psi gauge. Nitrogen for the fuel mixing: 310 psi regulated/350 psi gauge. Oxygen: 420 psi regulated/2400 psi gauge. Gauge pressures are approximate and quoted for a new tank/bottle.
- 12.) Follow this heating procedure until desired temperature is reached 800 K.

Pre-Experiment

- 1.) Make sure the nitrogen and oxygen bottles are connected to the manifold lines.
- 2.) Change reactor conditions temperature on the Experiment Design Excel spreadsheet. This is the temperature that the sample tip thermocouple is at.
The following “set” values will be from the Experimental Design Excel spreadsheet for the fuel calibration set point section.
- 3.) Read nitrogen set point.
- 4.) Turn on nitrogen flow controller red switch.
- 5.) Set proper flow for nitrogen.
- 6.) Turn air flow controller to off.
- 7.) Turn air flow controller red switch off.
- 8.) Read nitrogen fuel set from Experimental Design Excel spreadsheet.
- 9.) Set flow for nitrogen fuel.
- 10.) Turn on nitrogen fuel bead heaters.
- 11.) Pressurize the flow reactor to 8 atm.

Note: At this point in time only nitrogen is flowing through the reactor.

Clean/Calibrate the CO-CO₂ Detector

- 1.) Disconnect sample line here. (See Figure A.3, #1)
- 2.) Connect small nitrogen line to here. (See Figure A.3, #2)
- 3.) Open green valve on small nitrogen line. This will allow low pressure nitrogen to enter the analyzer.
- 4.) Press this button in slightly until the flow meter ball rises and the numbers on the display go to zero. (See Figure A.3, #3)
- 5.) Repeat this procedure 5 times.
- 6.) Turn green nitrogen small line valve off.
- 7.) Remove small nitrogen line.
- 8.) Reinstall sample line.
- 9.) Press this button in. (See Figure A.3, #4)
- 10.) Open air operated sample valve on the manifold by the FTIR. (See Figure A.4)
- 11.) Adjust flow meter ball to 3 by turning the flow adjusting valve. (See Figure A.3, #1)

Filling Syringe Pump with Fuel

- 1.) Find fuel flow rate. This is done by using the Total Hydrocarbon Analyzer or theoretically with the Experimental Design Excel spreadsheet.

- 2.) Calculate how much fuel (approximately 100 ml) is needed for the experiment. Possibly 2 hours worth.
- 3.) Attach funnel to syringe pump. Note: Funnel may be attached. (See Figure A.5)
- 4.) Check to see if old fuel is in pump.
- 5.) If yes, remove old fuel. Disconnect fuel line and press run on pump to remove fuel. Dispose of old fuel in an approved waste fuel container.
- 6.) Suck out remaining fuel with vacuum pump.
- 7.) Now perform the flush and suck method to clean syringe pump. Put fuel in the funnel. Press refill on pump. Stop when funnel is empty. Press run to empty fuel/flush. Suck remaining fuel/flush out with the vacuum pump. Repeat this 2-3 times or until all of the old fuel is removed.
- 8.) Fill pump with the correct amount of fuel.
- 9.) Change flow rate to around 10 ml/min.
- 10.) Purge pump by removing fuel line and press run until fuel starts to exit the pump.
- 11.) Connect fuel line to the pump.
- 12.) Make sure black outlet valve is opened on pump.
- 13.) Set the appropriate fuel flow rate.
- 14.) Open oxygen bottles and oxygen manifold valve. Note: 420 psi regulated/2400 psi gauge (new bottle).
- 15.) Open SV-4 oxygen switch.

- 16.) Adjust oxygen flow rate with flow controller. The appropriate value is from the Experimental Design Excel spreadsheet.
- 17.) Readjust pressure of the flow reactor if necessary to 8 atm.
- 18.) Readjust the nitrogen flow rate to the experimental set value from the Experimental Design Excel spreadsheet.
- 19.) Press run on the fuel syringe pump. Make sure the fuel flow rate is properly set. Let the fuel flow enough time to clear the fuel line from any fuel that was in it from previous experiments. Approx. 10 minutes. Make sure the reactor pressure is set to 8 atm.

Software Preparation

- 1.) Make a new folder. Copy the old data from previous experiment and paste into new folder.
- 2.) Open PFR control. This is a LabView program.
- 3.) Click on Initialize System tab. Click run. No errors should be present.
- 4.) Click Probe Control tab. Move bar to on. Click run. Note: Green online light should illuminate.
- 5.) Move probe into design point. First notice the temperature of the sample probe. It should be around 800 K. Type this temperature into the Experimental Design Excel spreadsheet's reactor conditions. Then note the probe position's experimental set point. The initial position of the probe should be greater than this set point. For example, if the spreadsheet calculates the position to be 31.275, put the probe at 32.0. To move the probe,

set slider bar to move, in, adjust how much to move, then click run. Note:

Wait until probe stops moving; i.e. this is when the numbers on the Daedel positioning controller stops changing.

- 6.) Click Probe Automove tab. Enter the residence time (from the reactor conditions), total flow rate (make sure this is total flow rate), and the maximum allowed position change (set to 2 at this time). Click “check mark” to enter values.
- 7.) Click on the Flow Controller Setpoints tab. Adjust flow controller setpoints. Click “check mark” to enter values.
- 8.) Go under Operate menu. Check “log data at completion” is selected.
- 9.) Go under Operate menu. Data logging. Log. If nothing shows up, clear the log file binding. Then repeat 9.
- 10.) Click rawlog.txt. Make sure you are in the right folder. Then delete. Keep this name, then save.
- 11.) Click Probe Automove tab.
- 12.) Make sure syringe pump is injecting fuel into the reactor and that sufficient time has passed to allow old fuel to be removed from the line.
- 13.) Before continuing, make sure the PFR sample probe temperature has stabilized. Also, make sure the nitrogen fuel bead heaters are up to temperature (approximately 400 °C or desired temperature).
- 14.) Click continuous run in LabView PFR Auto Control.

- 15.) Change the maximum allowed position change to .2, once the position change fluctuation decreases (this may take 10 seconds or so.) Click “check mark” to enter the number.
- 16.) Turn off the 10 kW and 3 kW heaters. Also, make sure the inlet, test and outlet bead heaters are off. Note: Leave the nitrogen fuel heater and breaker on.
- 17.) Control the pressure of the reactor around 8 atm during the controlled cool down experiment.

Data Collection After Controlled Cool Down

- 1.) Stop the fuel flow.
- 2.) Turn off nitrogen fuel heater and nitrogen fuel flow.
- 3.) Turn off oxygen flow controller, then SV-4 oxygen switch.
- 4.) Stop PFR Auto Control.
- 5.) Open PFR Datalogger. (Currently on the desktop) Click run. Click convertedlog.csv (make sure this is the right file and folder! This is the file that was copied in step 1 of the Software Preparation.) Delete this file. Use name. Click OK. Click large red Done button to finish.
- 6.) Open convertedlog.csv. Select all of the data. Copy.
- 7.) Open CCD Data Analysis X.xls. Go to log on the lower tab in Excel. Paste this data over the old. Click save as, save.
- 8.) Close PFR Data Logger.

Shut Down Procedure

- 1.) Move probe back to the home position. Click Probe Control in PFR auto control program. Move slider to off, click run. Move slider to on, click run. Make sure probe positioning table has stopped. The Daedel controller should read zero. Move to off, click run.
- 2.) Close valve for fuel on the syringe pump.
- 3.) Turn off transfer sample line heater.
- 4.) Turn off the heater for the probe/transfer sample line, grey Variac.
- 5.) At about 250 °C, lower the pressure to 1 atm. Do this gradually!
- 6.) Turn off Pacesetter at less than 100 °C. The Pacesetter can be left on for several hours, if needed. Make sure the blue water supply valve is turned off.
- 7.) Switch nitrogen SV-2 switch off. Turn nitrogen flow controller setpoint to zero. Turn SV-5 air switch on. Adjust air with the flow controller and set to 40.
- 8.) Make sure the date and name are changed on the Experimental Design Excel spreadsheet. Click save as, save.
- 9.) Turn pressure building valve off on the nitrogen tanks.
- 10.) Turn the nitrogen tanks and manifold off.
- 11.) Turn the oxygen tanks and manifold off.
- 12.) Once the reactor cools down to around room temperature 24 °C, turn off the airflow SV-5 switch.

A.1 Typical PFR Heater Set-Point Schedule

Table A.1 is a typical 10 kW and 3 kW heater schedule. At the start of each PFR experiment the 10 kW is set to 40 °C and the 3 kW set to 6 %. The 10 kW heater is increased by 20 °C every 15-20 minutes and the 3 kW increased every 40-50 minutes. Having the warm-up phase follow this schedule, both heaters will reach their safe maximum operating temperature at approximately the same time. The maximum safe operating parameter of the 10 kW heater is 427 °C and the 3 kW heater is 70 %. Each of the bead heaters are turned on separately throughout the warm up phase. The Pacesetter sample probe water temperature controller can be turned on when the temperature of the PFR is at 100 °C. The Pacesetter is turned on at 100 °C to prevent the water inside the sample probe from boiling. The nitrogen is introduced into the PFR at around 800 K and the nitrogen bead heaters are turned on. When the nitrogen is first introduced, the sample probe temperature will drop. However, after a few minutes from when the bead heaters are turned on, the temperature will start to increase. It is necessary to follow this schedule as closely as possible so that isothermal conditions are met.

Table A.1: Typical set-points for PFR heaters during warm-up phase.

Time	10 kW (°C)	3 kW (%)	Sample Temp (°C)	Notes
8:08 am	40	6	17.9	
8:28	60	---	26.5	
8:45	80	11	28.2	
9:03	100	---	37.0	
9:24	120	14	42.3	Inlet 1 on
9:44	140	---	82.9	Pacesetter on
9:58	160	18	104.4	Pacesetter set to 50°C
10:14	180	25	126.2	
10:41	200	---	156.0	
11:11	220	32	179.4	Test 1 on
11:28	---	38	202.0	
11:52	240	---	232.0	Inlet 2 on
12:14 pm	260	44	286.7	Outlet 1 on
12:39	280	50	340.0	
1:00	300	---	373.8	Test 2 on
1:15	320	55	410.5	Inlet 3 on
1:40	340	---	435.2	
2:02	360	60	455.2	Pacesetter set to 90°C
2:20	380	61	473.6	
2:42	400	62	494.1	
3:03	410	---	509.9	
3:19	---	68	515.4	
3:25	415	69	533.4	Nitrogen on & heaters
3:52	---	70	536.5	
4:45	---	---	557.6	Fuel sample stored
5:00	410	66	562.9	

A.2 Pictures of Selected PFR Facility Components



Figure A.1: Picture of PFR controllers with arrows identifying main power supply control switches.



Figure A.2: Picture of PFR computer with the arrow identifying the PFR mass flow controller.

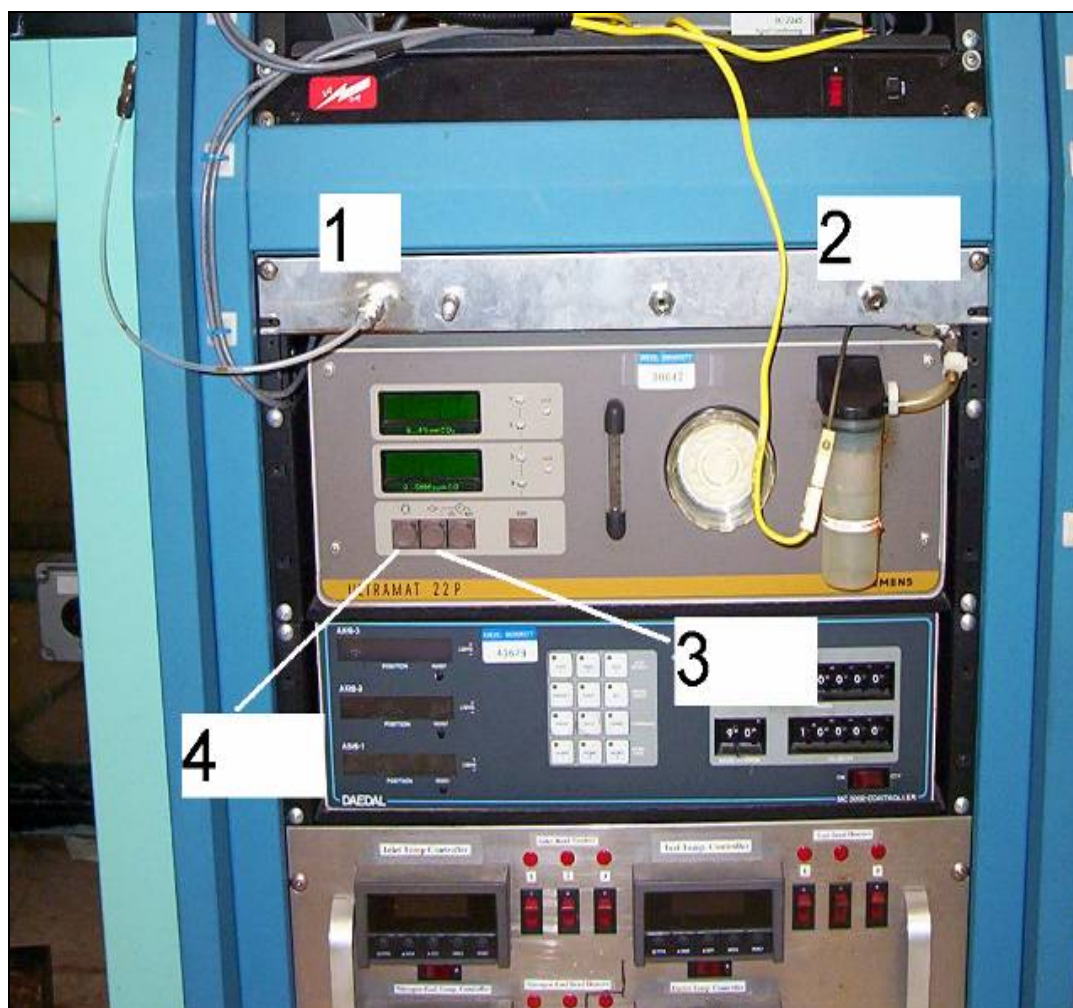


Figure A.3: Picture of sample connections to the CO and CO₂ detector, and calibration controls.

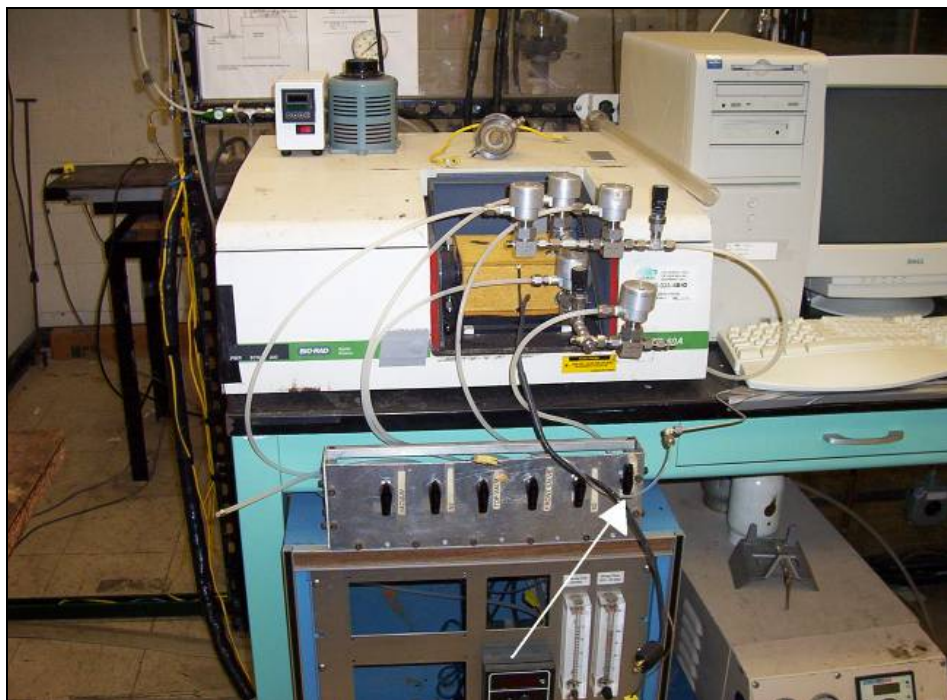


Figure A.4: Picture of air operated sample valve identified with arrow.

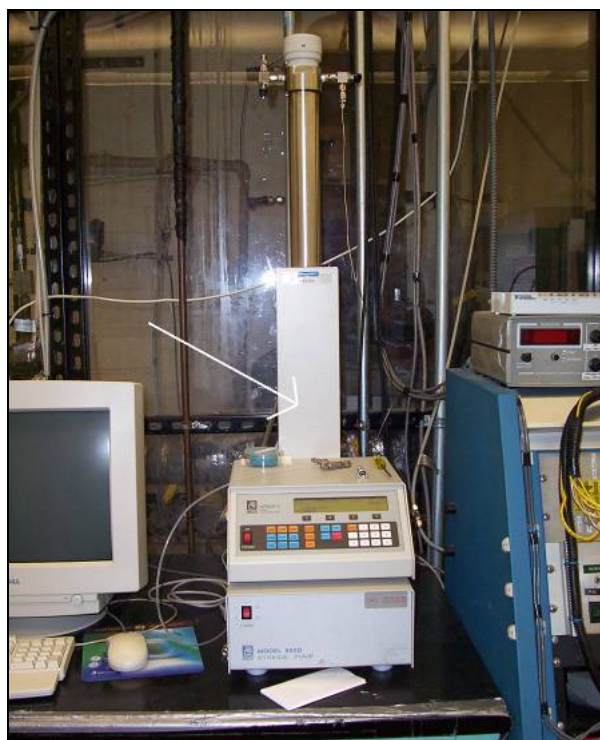


Figure A.5: Picture of the ISCO Model 500D fuel syringe pump.

